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# Prediction of Multiphase Flow Pattern inside a 3D Bubble Column Reactor Using a Combination of CFD and ANFIS

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## Abstract

This work presents a combination of Computational Fluid Dynamics (CFD) and Adaptive Network-based Fuzzy Inference System (ANFIS) developed for flow characterization inside a cylindrical bubble column reactor. An attempt has been made to predict the liquid flow pattern and gas dynamics for various ring sparger diameters (i.e., 0.07-0.16m) and bubble column heights. Gas hold-up, Turbulent Kinetic Energy (TKE) and axial liquid velocity are the output parameters predicted by using ANFIS method with respect to sparger diameter, axial coordination and radial coordination. Various architectures of the ANFIS method were constructed in order to achieve an accurate prediction model of the liquid flow behavior and gas dynamics inside the bubble column. ANFIS approaches were trained and tested by using CFD simulation results. The performance of the ANFIS approaches was examined by comparing the root mean square error and correlation coefficient values of the prediction models. The CFD simulation results are validated with existing experimental and numerical data and mathematical correlations. Both CFD simulation and ANFIS prediction results show that ring sparger diameter significantly changes the liquid flow pattern and gas dynamics, resulting different amount of the gas inside the column. Different ANFIS structures were selected for precisely estimation of gas hold-up, TKE and axial liquid velocity. Eventually, the mathematical correlations of the proposed ANFIS approaches are presented with correlation coefficients of 0.9717, 0.9917 and 0.9877 for gas hold-up, turbulent kinetic energy and axial liquid velocity prediction models. Hence, ANFIS approach is able to provide a prediction of the 3D bubble column hydrodynamics in a continuous domain.

**Keywords:** Bubble column reactor; Numerical method; Soft computing; CFD; ANFIS;  
Multiphase flow

34 **Nomenclature**

$C_D$	Drag force coefficient (-)
$C_{TD}$	Turbulent dispersion coefficient (-)
$C_{\varepsilon 1}$	Model parameter in turbulent dissipation energy equation (-)
$C_{\varepsilon 2}$	Model parameter in turbulent dissipation energy equation (-)
$C_\mu$	Constant in $k$ - $\varepsilon$ model (-)
$C_{\mu, BI}$	Constant in bubble induced turbulence model (-)
$d_B$	Bubble diameter (m)
$d_0$	Sparger hole diameter (m)
$D$	Diameter of the column (m)
$D_S$	Sparger diameter (m)
$g$	Gravitational constant ( $m/s^2$ )
$G$	Generation term ( $kg/m^2 s^2$ )
$H$	Height (m)
$k$	Turbulent kinetic energy per unit mass ( $m^2/s^2$ )
$M_I$	Total interfacial force acting between two phases ( $N/m^3$ )
$M_D$	Drag force ( $N/m^3$ )
$P$	Pressure ( $N/m^2$ )
$r$	Radial distance (m)
$R$	Column radius (m)
$Re_B$	Reynolds number ( $= d_B V_S / \nu$ ) (-)
$V_G$	Superficial gas velocity (m/s)
$V_y$	Axial liquid velocity (m/s)
TKE	Turbulent kinetic energy
MF	Membership function
RMSE	Root mean square error

**Greek Symbols**

$\varepsilon$	Turbulent energy dissipation rate per unit mass ( $m^2/s^3$ )
$\varepsilon$	Fractional phase hold-up (-)
$\bar{\varepsilon}$	Average fractional phase hold-up (-)
$\mu$	Molecular viscosity (Pa s)
$\mu_{BI}$	Bubble induced viscosity (Pa s)
$\mu_{eff}$	Effective viscosity (Pa s)
$\rho$	Density ( $kg/m^3$ )
$\mu_T$	Turbulent viscosity (Pa s)
$\sigma$	Surface tension (N/m)
$\sigma_\varepsilon$	Prandtl number for turbulent energy dissipation rate (-)
$\sigma_k$	Prandtl number for turbulent kinetic energy (-)
$\tau_k$	Shear stress of phase k (Pa)
$\varepsilon_g$	Air fraction / Gas hold-up

**Subscripts**

$G$	Gas phase
$L$	Liquid phase

35  
36  
37

## 38        **1 Introduction**

39 Bubble column reactors are extensively used in several multiphase reaction processes within  
40 industrial applications such as chemical, biochemical and wastewater treatment [1-7]. They are  
41 preferred over other reactors due to high energy efficiency, low design costs, simple structure  
42 and operation [1, 3]. Furthermore, they have a good performance in phase mixing characteristics  
43 and heat and mass transfer (large interfacial areas) [1, 3, 8, 9]. They are often cylindrical or  
44 rectangular, including gas sparger (gas distributor) at the bottom. The sparger produces bubbles  
45 in a continuous liquid phase (stationary or flowing) or a liquid–solid suspension [1, 3, 10-15].

46 Design, optimization and manufacturing of these reactors highly depend on the complex  
47 dynamics of gas bubble interaction, liquid flow pattern and prevailing gas and liquid regime (i.e.,  
48 homogeneous or heterogeneous) [3, 11-13, 16-20]. One of the main challenges in enhancing the  
49 design and manufacturing of the bubble column is to properly predict and measure the  
50 hydrodynamics properties, while the complex behavior of the gas and liquid movement,  
51 including the interaction between bubbles are unavoidable [1, 3, 11, 12, 15, 21, 22]. Several  
52 experimental methods such as Particle Image Velocimetry (PIV), Laser Doppler Anemometry  
53 (LDA) and Radioactive Particle Tracking have been developed in recent years to develop an  
54 insight into this arguably complex and nonlinear behavior of gas and liquid dynamics,  
55 particularly liquid flow pattern and amount of gas inside bubble column reactors [23-26].

56 Apart from experimental techniques, many CFD approaches and mathematical calculations are  
57 available to predict bubble column hydrodynamics [3, 7, 13-15, 27-29]. Nowadays, high  
58 performance computers have enabled the use of detailed mathematical and computational  
59 approaches to study the liquid flow pattern and gas dynamics on a feasible time span and space.  
60 There are two main CFD approaches i.e., the Eulerian–Eulerian and Eulerian–Lagrangian to  
61 model multiphase fluid flow [3, 6, 10, 13-15, 19, 27-30]. In the Eulerian–Lagrangian approach  
62 (Discrete particle model), each bubble is separately tracked inside the bubble column by solving  
63 forces acting on the bubbles, while continuum description is considered for the liquid phase [6].  
64 In this approach, the interaction between bubbles i.e., coalescence, break-up and collisions can  
65 be observed. However, this framework is limited for large bubble columns with high number of  
66 bubbles due to solving more equations in large domains. On the other hand, Eulerian–Eulerian  
67 approach (the two fluid model), considers gas bubbles and liquid in the Eulerian framework as  
68 two interpenetrating fluids. Unlike the discrete particle model, the Eulerian framework is an

69 appropriate method to solve the large bubble column with high superficial gas velocity,  
70 particularly in industrial bubble column reactors [13-15, 22, 29].

71 Although several experimental, numerical and mathematical methods have been used to measure  
72 and estimate the flow pattern and bubbles dynamics [3, 5-7, 9, 31-39], there are some difficulties  
73 to completely predict the liquid flow pattern and gas dynamics (bubble coalescence, break-up,  
74 velocity, shape, size and gas hold-up) at each point of 3D bubble column reactors when the  
75 operation conditions (i.e., superficial gas velocity, column dimensions, gas and liquid properties  
76 and sparger parameters), flow regime and operation time change. For instance, measuring the  
77 fluid flow parameters inside the 3D bubble column reactor during experiment is extremely  
78 expensive and required much measurement equipment. Computation time and computer  
79 capability are the major limitations of the computational approaches in numerically simulation of  
80 the large bubble column and various operation conditions. Because of these limitations, soft  
81 computing methods have been developed to estimate the bubble column hydrodynamics in  
82 various conditions that have not been simulated or experimented at every point of the bubble  
83 column [37].

84 There are several soft computing techniques (e.g., neural networks, Support vector machines,  
85 evolutionary algorithms, and adaptive neuro fuzzy inference system) proposed in many studies to  
86 estimate phenomena behaviour in the real life applications [37, 40-47]. Among these techniques,  
87 Adaptive Network-based Fuzzy Inference System (ANFIS) has attracted researchers because of  
88 its ability to learn complex relationships and its vast application has been illustrated in numerous  
89 studies [41, 48-50]. The accuracy of the ANFIS approach can be altered by changing prediction  
90 model structure and adapted on the basis of the relationship complexity [40, 41, 51-58]. ANFIS  
91 method can use either simulation or experimental results as training data to learn the phenomena  
92 behavior. An appropriate set of training data is required to successfully train ANFIS model.

93 Azwadi et al.[41] used CFD results for training ANFIS method to estimate the temperature and  
94 flow fields in a 2D lid-driven cavity. They found that the result of ANFIS method is in good  
95 agreement with temperature and flow field obtained by CFD simulation. Recently, Pourtousi et  
96 al. [37] employed this methodology to predict multiphase flow inside a bubble column reactor.  
97 They utilized bubble column hydrodynamics data (i.e., liquid velocity components, turbulent  
98 kinetic energy and gas hold-up), obtained by CFD (Eulerian method) simulation, at the bubble  
99 column bulk region for ANFIS learning process. It was found that the combination of CFD and

100 ANFIS is a robust methodology to predict the bubble column hydrodynamics properties in a  
101 continuous domain. They showed that ANFIS method can be a favourable replacement with  
102 CFD simulation to predict the complex behaviour of multiphase flow inside the bubble column  
103 reactor when the flow regime is homogeneous.

104 In this study we develop the recent methodology (Pourtousi et al.'s research [37]) to propose an  
105 intelligent approach which is able to model multiphase flow inside the bubble column reactor for  
106 various sparger diameters. In addition, an attempt has also been made to improve the overall  
107 predictive capabilities of liquid flow pattern and gas hold-up using the combination of CFD and  
108 ANFIS methods. A new mathematical correlation is proposed to predict the bubble column  
109 hydrodynamics as the ring sparger diameters varied from 0.07 to 0.16m. The effect of ring  
110 sparger diameter on liquid flow velocity, turbulent kinetic energy and gas hold-up is investigated  
111 using ANFIS and CFD results. Various ANFIS structures were constructed to realize the most  
112 accurate structure for each output. The accuracy of all prediction models was compared by two  
113 common error evaluation formulas; root means square error and correlation coefficient. The  
114 results of selected ANFIS models were compared to the CFD simulation results to illustrate the  
115 capability of the ANFIS approach.

116

117

## 118 **2 Methodologies**

### 119 **2.1 Geometrical structure**

120 In this CFD simulation study, a 3D cylindrical bubble column reactor with 2.6m height  
121 and 0.288m diameter is used to produce the multiphase flow (bubbling process). The bubble  
122 column is filled with stationary water at ambient condition. At the bottom of the column, 20  
123 similar holes (sparger) whose diameter varied from 0.07 to 0.16 m are symmetrically defined a  
124 circle with diameter of 0.7m. The superficial gas velocity for all simulation cases are 0.005m/s,  
125 resulting in homogeneous flow regime with almost uniform bubble sizes, shapes and velocities.

126

127

128

129

### 130 **2.2 Combination of CFD and ANFIS methods**

131 The prediction of the 3D bubble column hydrodynamics is started by CFD simulation of  
 132 10 case studies with different sparger diameters. The CFD simulation results are employed to  
 133 train the ANFIS approaches. Fig. 1 portrays the schematic of prediction procedure of bubble  
 134 column using the combination of CFD and ANFIS.

### 135 2.3 CFD modeling

136 In the current CFD simulation study, the Euler–Euler approach as incorporated in the  
 137 commercial ANSYS CFX-13 software is employed to model the multiphase flow inside the  
 138 bubble column. The Eulerian framework of the gas and liquid flow represents a point volume  
 139 fraction for the gas and liquid individually. This method is based on the notion of pseudo-  
 140 continuum. Both the liquid and gas phases (multi-bubbles) are formulated in the Eulerian  
 141 framework as interpenetrating continua. The formulations of the Eulerian model, selected in the  
 142 present CFD study, are based on ensemble-averaged mass and momentum transport equations  
 143 for the gas and liquid separately. The continuity and momentum transfer equations for the Euler–  
 144 Euler multi-phase framework are represented respectively (solved for the gas and liquid phases  
 145 individually) as follows:

146 The mass conservation equation for both liquid and gas is shown as follows:

$$147 \frac{\partial}{\partial t}(\rho_k \epsilon_k) + \nabla(\rho_k \epsilon_k u_k) = 0 \quad (1)$$

148 where  $\epsilon_k$  and  $u_k$  indicate the volume fraction and average velocity of phase  $k$  respectively.

149 In the present numerical investigation, the control volume method is used to discretize the  
 150 conservation equations. There are several solution methods (such as finite difference [59],  
 151 Lattice Boltzmann [60-63], finite volume method [13-15, 27, 28], etc.) in the CFD to solve the  
 152 fluid flow problems. The most robust, reliable and the one, on which CFX is based, is called  
 153 finite volume discretization method. Based on the finite volume discretization method, the  
 154 momentum transfer formulation for multi-bubbles and liquid phases can be described as:

$$155 \frac{\partial}{\partial t}(\rho_k \epsilon_k u_k) + \nabla(\rho_k \epsilon_k u_k u_k) = -\nabla(\epsilon_k \tau_k) - \epsilon_k \nabla p + \epsilon_k \rho_k g + M_{I,K} \quad (2)$$

156 The right side of the momentum transfer formulation consists of the stress, pressure  
 157 gradient, gravity and the momentum interfacial exchange between gas bubbles and liquid. In this  
 158 equation, the stress term of phase  $k$  is represented as follows:



$$159 \quad \tau_K = -\mu_{eff,k} (\nabla u_k + (\nabla u_k)^T) - \frac{2}{3} I(\nabla u_k) \quad (3)$$

160 where  $\mu$  indicates the liquid phase effective viscosity, comprising of three terms: molecular  
161 viscosity, turbulence viscosity and viscosity based on bubble induced turbulence.

$$162 \quad \mu_{eff,l} = \mu_L + \mu_{T,L} + \mu_{BI,L} \quad (4)$$

163 The effective gas viscosity is formulated based on the effective viscosity of liquid and it can be  
164 described as follows:

$$165 \quad \mu_{eff,G} = \frac{\rho_G}{\rho_L} \mu_{eff,L} \quad (5)$$

166 In the current CFD simulation study, the model of Sato and Sekoguchi is employed for  
167 the extra term due to bubble induced turbulence, containing a constant value of  $C_{\mu,BI} = 0.6$ . The  
168 viscosity due to the turbulence induced by the gas bubble flow has been described by [64, 65].  
169 They demonstrated a model taking account the turbulence induced by bubble agitation inside the  
170 liquid phase. In general, to predict momentum of bubble flow it is crucial to describe the  
171 turbulent structure of the continuous liquid phase, which may result in how to describe the  
172 contribution of bubble existence to the flow characteristics. Sato and Sekoguchi [64], reported  
173 that the turbulent shear stress in bubble flow is affected by two terms. Firstly, the inherent liquid  
174 turbulence which is independent of relative motion of bubbles in the liquid phase. Secondly, The  
175 additional liquid turbulence term, producing by bubble agitation (bubble motion).

$$176 \quad \mu_{BI,L} = \rho_L C_{\mu,BI} \epsilon_G d_B |u_G - u_L| \quad (6)$$

177 The last term in the momentum transfer equation is the total interfacial force. This term can be  
178 described as follows:

$$179 \quad M_{I,L} = -M_{I,G} = M_{D,L} + M_{TD,L} \quad (7)$$

180 The total interfacial forces, illustrated above, indicate the drag and turbulent dispersion  
181 force when the lift and virtual mass are neglected. The interphase momentum transfer between  
182 gas bubble and liquid phase due to drag force is shown as follows:

$$183 \quad M_{D,L} = -\frac{3}{4} \epsilon_G \rho_L \frac{C_D}{d_B} |u_G - u_L| (u_G - u_L) \quad (8)$$

184 where the  $C_D$  and  $d_B$  are the drag coefficient and bubble diameter respectively. In general, the  
185 drag coefficient and bubble diameter can be assumed as a constant value due to uniform



186 behaviour of bubble size and shape in homogeneous regime. The drag coefficient and bubble  
 187 diameter are selected as 0.44 and 4mm diameter, respectively, based on the literature suggestions  
 188 [3, 13, 14, 37], experimental observation and numerical setting of Pflieger and becker's study  
 189 [13].

190 Turbulent dispersion force model is used for current CFD investigations based on  
 191 literature studies [3, 15, 22, 37, 66] to improve the flow field prediction towards the walls. This  
 192 model, formulated by Lopez de Bertodano [67], is on the basis of the analogy with molecular  
 193 movement and interaction. It approximates a turbulent diffusion of the bubbles by the liquid  
 194 eddies and can be described as:

$$195 \quad M_{TD,L} = -M_{TD,G} = -C_{TD}\rho_L K \nabla \epsilon_L \quad (9)$$

196 where  $k$  and  $C_{TD}$  indicate the liquid TKE and turbulent dispersion coefficient. Various values of  
 197 turbulent dispersion coefficient have been recommended in the previous studies [3, 15, 22, 37,  
 198 66]. In our simulation setting, turbulent dispersion coefficient of 0.3 is used based on the  
 199 sensitivity study [3, 15, 37, 66]. The sensitivity study has been carried out for turbulent  
 200 dispersion coefficients from 0.2 to 0.5 and the result shows a small difference in results of flow  
 201 pattern and gas hold-up, while the value of 0.3 results in marginal improvement of flow pattern  
 202 results.

203 In addition to interfacial forces, a proper selection of turbulence model is necessary to  
 204 appropriately predict the bubble column hydrodynamics [3, 12-16, 22, 29, 37, 38, 66]. For the  
 205 disperse bubbly phase a zero equation turbulence model is used. However, the standard  $k-\epsilon$   
 206 model is applied for the continuous phase which have been used and recommended in prior CFD  
 207 studies due to obtaining average results, simplicity and low computation time [3, 12-16, 22, 27-  
 208 29, 37, 66, 68]. As  $k-\epsilon$  is employed for turbulence modelling, the turbulent eddy viscosity is  
 209 calculated using the standard  $k-\epsilon$  turbulence model, where  $k$  represents the turbulent kinetic  
 210 energy and  $\epsilon$  its dissipation rate in the liquid phase.  $k$  and  $\epsilon$  determine the energy in turbulence  
 211 and the scale of the turbulence, respectively. The turbulent eddy viscosity can be defined as  
 212 follows:

$$213 \quad \mu_{T,L} = \rho_L C_\mu \frac{K^2}{\epsilon} \quad (10)$$

214 The turbulent kinetic energy ( $k$ ) and its energy dissipation rate ( $\epsilon$ ) are calculated based on  
 215 the following governing equations:

$$216 \quad \frac{\partial}{\partial t}(\rho_L \epsilon_L K) + \nabla(\rho_L \epsilon_L u_L K) = -\nabla(\epsilon_L \frac{\mu_{eff,L}}{\sigma_K} \nabla_K) + \epsilon_L (G - \rho_L \epsilon) \quad (11)$$

$$217 \quad \frac{\partial}{\partial t}(\rho_L \epsilon_L \epsilon) + \nabla(\rho_L \epsilon_L u_L \epsilon) = -\nabla(\epsilon_L \frac{\mu_{eff,L}}{\sigma_\epsilon} \nabla_\epsilon) + \epsilon_L \frac{\epsilon}{K} (C_{\epsilon 1} G - C_{\epsilon 2} \rho_L \epsilon) \quad (12)$$

218

219 Being  $k$  and  $\epsilon$  calculated from their conservation equations. The  $k$ - $\epsilon$  model is applied, in this  
 220 work, with its standard constants values (model parameters):  $C_\mu = 0.09$ ,  $\sigma_k = 1$ ,  $\sigma_\epsilon = 1$ ,  $C_{\epsilon 1} = 1.44$ ,  $C_{\epsilon 2} =$   
 221  $1.92$ . These constants, although not universal, are commonly used in the case of single-phase flow [19,  
 222 69]. The selection of these values based on recommendation of prior numerical studies [13, 15, 22, 37,  
 223 70]. The term  $G$  indicates the production of turbulent kinetic energy and can be represented as:

$$224 \quad G = \tau_L : \nabla u_L \quad (13)$$

225

### 226 2.3.1 Grid

227 For meshing the cylindrical bubble column, the hexahedral grid is used throughout the column  
 228 which is almost similar with study of Boutet et al. [71]. The circular cross section of bubble  
 229 column is non-uniformly meshed, while the axial length of column is equally divided into 60  
 230 grids section. Fig. 2 shows a typical radial and axial grid layout for the 3D bubble column  
 231 structure, containing 40500 grids. On average, this grid structure has the orthogonal quality of  
 232 0.67, skewness of 0.62 and aspect ratio of 3.1. In addition, two denser meshes (i.e., 59000 and  
 233 82320 elements) with similar structure are used for grid sensitivity study.

234

### 235 2.3.2 Boundary conditions

236 Instead of modelling the exact ring sparger for inlet condition, the mass source point is  
 237 used for each sparger hole, calculating based on superficial gas velocity. At the top surface of the  
 238 bubble column (outlet), a degassing boundary condition is treated, resulting in no penetration and  
 239 slip condition for the liquid phase and an outlet for bubbles. In this case, the pressure remains  
 240 variable on the top of the column, describing the various surface heights at different bubble  
 241 column locations. On the walls, a no slip and free slip boundary conditions are used for the liquid  
 242 and gas phases respectively. Considering the free slip boundary condition for the gas phase in  
 243 Eulerian method describes the interaction between multi-bubbles and solid walls. As bubbles  
 244 move towards the walls without friction, they experience no interaction (freely movement) and

245 the direct contacts between multi-bubbles and walls can be considered negligible. In this way,  
 246 the velocity component parallel to the wall has a finite value, while both the velocity normal to  
 247 the wall and the wall shear stress are set to zero.

248

### 249 2.3.3 Numerical methods

250 In order to mathematically discretize the conversation equations, the control volume  
 251 technique is implemented throughout the 3D cylindrical bubble column. The velocity–pressure  
 252 connection is achieved using SIMPLEC procedure. The high order differencing schemes of total  
 253 variation diminishing (TVD) is used to decrease numerical diffusion in current CFD simulation  
 254 study. The bubbling process is simulated for 1400s and the results are time averaged over last  
 255 1300s.

256

### 257 2.4 Adaptive-Network-based Fuzzy Inference System (ANFIS)

258 ANFIS is an inference fuzzy system to accurately predict the behavior of complex and  
 259 nonlinear systems [40, 41, 47, 53, 54, 72]. There are three different types of fuzzy reasoning in  
 260 which Takagi and Sugeno proposed if-then rules are implemented in ANFIS structure [73]. Fig.  
 261 3 shows the structure of the employed ANFIS method on predicting the hydrodynamic  
 262 characteristics in the 3D bubble column. In this paper (three inputs, sparger diameter, radial  
 263 coordination and axial coordination) are taken to obtain the gas hold up, TKE and axial liquid  
 264 velocity as output (see Fig. 3). The inputs are divided into various numbers of membership  
 265 functions (MFs) in first layer. The incoming signals from first layer are multiplied according to  
 266 AND rule as the node function for the second layer. For instance, the function of the  $i^{\text{th}}$  rule is as,

$$267 \quad w_i = \mu_{A_i}(D_s) \times \mu_{B_i}(x) \times \mu_{C_i}(H) \quad (14)$$

268 where  $w_i$  is out coming signal of second layer's node and  $\mu_{A_i}$ ,  $\mu_{B_i}$  and  $\mu_{C_i}$  are incoming signals  
 269 from implemented MFs on inputs, sparger diameter ( $D_s$ ), axial coordination ( $H$ ) and radial  
 270 coordination ( $x$ ), to second layer's node.

271 In layer three, the relative value of firing strength of each rule is calculated. This value  
 272 equals to the weight of each layer over the total amount of all rules' firing strengths:

$$273 \quad \bar{w}_i = \frac{w_i}{\sum_{i=1}^n (w_i)} \quad (15)$$

274 where  $\bar{w}_i$  is called normalized firing strengths. The fourth layer applied the function of a  
 275 consequence if-then rule proposed by Takagi and Sugeno [73]. Thus, the node function can be  
 276 described as:

$$277 \quad \bar{w}_i f_i = \bar{w}_i (p_i D_s + q_i x + r_i H + s_i)$$

278 where  $p_i$ ,  $q_i$ ,  $r_i$  and  $s_i$  are the if-then rules' parameters and called consequent parameters. All  
 279 incoming signals from layer four are aggregated to obtain the model output represents the  
 280 estimation result.

281 A hybrid learning algorithm is utilized to update the parameters in which MFs parameters are  
 282 updated by gradient descent method and consequent parameters are updated by Least Square  
 283 Estimate (LSE) method.

284

#### 285 2.4.1 Membership function selection

286 One of the main aspects of this research is the investigation of the best type and number of  
 287 MFs for all inputs in terms of the Root Mean Square Error (RMSE) and Correlation Coefficient  
 288 (CC). The equation of RMSE can be defined as:

$$289 \quad RMSE = \sqrt{\frac{1}{N} \sum_{i=1}^N (Actual\ Output - Estimated\ Output)^2} \quad (15)$$

290 where  $N$  is the number of testing sets.

291 The equation of CC that provides the relationship strength between prediction and CFD  
 292 simulation results is as:

$$293 \quad CC = \frac{\sum_{i=1}^N (y_{CFD(i)} - y_{CFD(m)})(y_{pre(i)} - y_{pre(m)})}{\sqrt{\sum_{i=1}^N (y_{CFD(i)} - y_{CFD(m)})^2 \sum_{i=1}^N (y_{pre(i)} - y_{pre(m)})^2}} \quad (16)$$

294 where  $y_{CFD(i)}$  and  $y_{CFD(m)}$  are the CFD result of individual data and the mean value of CFD  
 295 results.  $y_{pre(i)}$  and  $y_{pre(m)}$  are the ANFIS prediction result of each data and the mean value of the  
 296 prediction results.

297 In this aim, Bell-shaped, Gaussian and Sigmoidal MFs, and 27 combinations of number  
 298 of MFs are employed to compare the estimation output errors. Table 1 portrays the equation of  
 299 utilized MFs in the ANFIS model and Tables 2 and 3 depict the RMSE and CC values of various  
 300 employed ANFIS structures.

301

302 Gaussian MF has two parameters ( $c$  and  $\sigma$ ), Bell-shaped MF has three parameters ( $a$ ,  $b$ ,  
303 and  $c$ ), and Sigmoidal MF has four parameters ( $a_1$ ,  $a_2$ ,  $c_1$ , and  $c_2$ ). These parameters are called  
304 premise parameters. Numbers of MF are varied from 2 to 4 for each input and ANFIS structures  
305 are configured from 2-2-2 configuration, which means 2 MFs for sparger diameter ( $D_s$ ), axial  
306 coordination ( $H$ ) and radial coordination ( $x$ ), to 4-4-4 configuration.

307

308 As seen in Tables 2 and 3, the most accurate ANFIS structures in terms of RMSE and CC  
309 values are 4-4-4 configuration with Sigmoidal MF for gas hold-up, 4-4-4 configuration with  
310 Bell-shaped MF for TKE and axial liquid velocity. Complexity of the model is another  
311 considerable parameter in selecting best ANFIS structure. In this aim, the total number of  
312 parameters, comprising premise and consequent parameters, is obtained for each model. This  
313 value is varied from 44 to 304 parameters.

314 Figs. 4, 6 and 8 portray gas hold-up, TKE and axial liquid velocity RMSE values of ANFIS  
315 structures in terms of number of parameters (premise and consequent) while Figs. 5, 7 and 9  
316 show gas hold-up, TKE and axial liquid velocity CC values of ANFIS structures in terms of  
317 number of parameters. As seen in Fig. 4, RMSE results of gas hold-up illustrate an error  
318 reduction when the number of parameters increases. The RMSE results are distributed in two  
319 regions; high and low RMSE values. Different ANFIS configurations may have similar number  
320 of parameters which represents the complexity of the model.

321

322 As instance, ANFIS models with configuration of 2-3-4, 2-4-3, 3-2-4, 3-4-2, 4-2-3, and 4-  
323 3-2 have 114, 123, and 132 parameters for Gaussian, Bell-shaped, and Sigmoidal MFs,  
324 respectively. Type of MF has a slight influence on the gas hold-up prediction accuracy of gas  
325 hold-up while the number of parameters has a key role on the model precision. ANFIS models  
326 with configurations of 2-3-4, 2-4-3, 4-2-3, and 3-2-4 are much more accurate than models with  
327 configurations of 3-4-2 and 4-3-2. This comparison shows that radial coordination ( $x$ ) needs to  
328 be split into more spaces by means of more MFs for  $x$ . Hence, ANFIS model with 4-4-4  
329 configuration and Sigmoidal MF is selected as the best model to predict gas hold-up by  
330 comparing the RMSE and CC values and the influence of the ANFIS model configuration on the  
331 prediction accuracy.

332

333 Fig. 6 depicts the relationship between TKE RMSE values of diverse ANFIS models and  
334 numbers of parameters. In contrast with error result of gas hold-up prediction, TKE RMSE  
335 results are close together and reduction rate is lesser. Thus, a simpler ANFIS structure with less  
336 number of parameters can be chosen due to its simplicity and low RMSE value. ANFIS model  
337 with configuration of 4-4-4 has 280, 292, and 304 parameters for Gaussian, Bell-shaped, and  
338 Sigmoidal MFs while ANFIS model with configuration of 4-4-3 for Bell-shaped MF has 225  
339 parameters. The RMSE values of these ANFIS models are close together. Therefore, the ANFIS  
340 model with lowest number of parameters can be selected as the best model to predict TKE. Fig. 7  
341 illustrates that the ANFIS model with configuration of 4-4-3 and Bell-shaped MF has an  
342 appropriate performance in terms of CC value and is selected as the best ANFIS model for  
343 estimation of TKE.

344  
345 Axial liquid velocity is another parameter that has been predicted by ANFIS method. Fig. 8 shows  
346 the RMSE values of diverse ANFIS structures versus number of parameters. As can be seen, the error  
347 result of utilizing Sigmoidal MF is higher than that of utilizing Bell-shaped and Gaussian MFs. The  
348 RMSE and CC results of ANFIS model with 2-4-4 configuration and Gaussian MF are close to those of  
349 models with higher complexity such as 3-4-4 configuration with Bell-shaped MF, 4-4-4 configuration  
350 with Gaussian and Bell-shaped MFs (see figs. 8 and 9). Hence, the best model in terms of both simplicity  
351 and accuracy is ANFIS model with 2-4-4 configuration and Gaussian MF to predict axial liquid velocity.

352

353

354

### 355 **3 Results and discussion**

#### 356 **3.1 Validation of CFD method**

357 As the first step, it is important to establish the validity of the CFD model for prediction  
358 of the flow pattern and amount of gas inside the bubble column reactor. Therefore, a comparison  
359 has been made with the previous experimental and numerical data (i.e., Pflieger and Becker [13]  
360 and Diaz et al. [19]) and existing mathematical correlations (i.e., Joshi and Sharma [74], Kumar  
361 et al. [75], and Hughmark [76]). Excellent comparison is obtained between the current CFD  
362 estimation and previous studies for overall gas hold-up. For instance, Fig. 10 shows the overall  
363 gas hold-up against different superficial gas velocities for the present CFD simulation and  
364 previous studies. The overall gas hold-up inside the bubble column reactor linearly rises when

365 the superficial gas velocity increases. This figure shows that the present finding of gas hold-up is  
366 in satisfactory agreement with experimental and numerical studies of Pflieger and Becker [13]  
367 (especially experimental finding) and Diaz et al. [19] and mathematical correlations of Joshi and  
368 Sharma [74] and Hughmark [76], when the superficial gas velocity alters from 0.0015m/s to  
369 0.01m/s. However, the numerical study of Pflieger and Becker [13] overpredicts the gas hold-up  
370 almost for all superficial gas velocities.

371  
372 Fig. 11 shows the planer averaged gas hold-up versus bubble column heights for Grids 1  
373 and 3 and Pflieger and Becker's investigation [13]. In comparison to Pflieger and Becker's  
374 numerical study, current numerical results are in excellent agreement with experimental data,  
375 especially near the sparger. The Pflieger and Becker [13] showed an overprediction for gas hold-  
376 up at various column heights (particularly near the spargers). The figure shows that, towards the  
377 sparger, the numerical study of Pflieger and Becker [13] could not accurately estimate gas hold-  
378 up (over prediction), while the current CFD results are in good agreement with experimental  
379 data. In addition, the figure also illustrates that Grid 1 presents better agreement in comparison  
380 with Grid 3 in all column heights, particularly near the sparger.

381

382

### 383 3.2 Grid dependency

384 Three types of grids are used; Grid 1, Grid 2 and Grid 3 (mention in grid section), in  
385 order to investigate the effect of the mesh resolution on the results of flow pattern and amount of  
386 gas inside the bubble column. The CFD results of axial liquid velocity, based on three types of  
387 grids, are benchmarked against that from Pflieger and Becker experimental and numerical study  
388 [13]. Fig. 12 portrays the time averaged axial liquid velocity versus the normalized radial  
389 coordinate for Grids 1, 2, and 3 and experimental and numerical results of Pflieger and Becker at  
390 1.6m column height. According to the figure, Grid 1 shows that axial flow is upward in the  
391 central region of column with higher gas hold-up, while a downward counter flow is observed  
392 towards the wall region with low gas hold-up. This hold-up gradient creates the density  
393 difference for liquid circulation to take place. The point of flow reversal is clearly seen at a radial  
394 location of around  $r/R = 0.6-0.8$ m. In contrast, the results of Grids 2 and 3 show the asymmetric



395 liquid profile in which Grids 2 and 3 tend to move towards the right and left walls respectively.  
396 The results of Grid 1 also are in good agreement with numerical data of Pflieger and Becker [13].

397 In general, the grid dependency test shows a good agreement when the grid size increases  
398 (coarse mesh), while the numerical results with finer mesh differ more from experimental data of  
399 Pflieger and Becker [13] (see Figs. 11 and 12). Buwa et al. [69] and Pflieger and Becker [13] also  
400 reported that, as the grid size decreases, the agreement between numerical results of time  
401 averaged flow pattern and experimental finding deteriorates. Furthermore, Bech [77] showed  
402 that transient turbulence models produce new modes of instability in the plume oscillation when  
403 the grid size decreases. Based on results of grid dependency illustrated in Figs. 11 and 12, the  
404 coarse mesh, containing 40500 grids, is used for all simulation cases in this study.

### 405 **3.3 Time step optimization**

406 The influence of the time-step size on the flow pattern and gas hold-up results is also  
407 studied. In order to investigate the size of time step, the Courant–Friedrichs–Levy (CFL)  
408 number, is used. The CFL order number of one is necessary to resolve the transient bubbling  
409 process inside the column. The CFL number can be described as follows:

$$410 \quad \Delta t \leq \frac{\Delta y}{|v|} \quad (17)$$

411 where  $|v|$ ,  $\Delta y$  and  $\Delta t$  are the magnitude of the velocity vector's vertical component,  
412 characteristic dimension of the cell and time step, respectively. In order to study the effect of  
413 using different time steps on the accuracy of CFD results, time steps, ranging 0.1-0.01 are tested  
414 which results in small differences between results. In this study the time step of 0.1 is used for all  
415 CFD study. The selection of the time step is also evaluated by the CFL that the maximum CFL  
416 number must be less than one. Several investigations showed that when the CFL is smaller than  
417 1, the numerical method can accurately predict the liquid flow pattern and gas dynamics and  
418 further refining of this parameter does not lead to significant changes on the results. However,  
419 using CFL larger than 1 results in inaccurate prediction results [17, 29, 38, 78-81].

### 420 **3.4 Prediction evaluations and discussions**

421 In this section, an accuracy evaluation and comparison between CFD and ANFIS results  
422 are discussed. Three different ANFIS models are proposed to predict the gas hold-up, TKE and

423 axial liquid velocity individually. The results are divided into three portions: prediction  
424 evaluation, surface rules and mathematical models.

425 ANFIS model has been trained by using CFD simulation results. To verify the trained  
426 model, four axial coordinates, 0.867, 1.3, 1.733 and 2.167 m, are employed to examine the  
427 prediction result. Figs. 13-15 depict both CFD simulation and ANFIS prediction results for gas  
428 hold-up, TKE and axial liquid velocity in the bubble column. These figures illustrate the high  
429 capability of ANFIS method to model the characteristics of bubbly flow.

430 Inside bubble column reactors, the amount of gas depends upon the regime operation  
431 (i.e., homogeneous/heterogeneous) which turn depends upon column dimensions, pressure,  
432 temperature, superficial gas velocity and sparger design parameters. Fig. 13(a-d) shows gas  
433 hold-up results of CFD simulation and ANFIS prediction model at various bubble column  
434 heights (0.867, 1.3, 1.73 and 2.16 m) and sparger diameters (0.08m (Fig. 13(a)), 0.10m (Fig.  
435 13(b)), 0.12 m (Fig. 13(c)) and 0.14 m (Fig. 13(d))). According to this figure, both CFD and  
436 ANFIS methods show that gas bubbles tend to move towards the bubble column center due to  
437 centralized gas movement in the homogeneous regime. An increase in sparger diameter from  
438 0.08 to 0.14m causes a significant decrease in the magnitude of gas hold-up at the column center.  
439 Among all sparger diameters, the sparger diameter 0.14m produces almost flatter gas hold-up  
440 profile inside the bubble column reactor. Both CFD and ANFIS methods show that the gas hold-  
441 up profile is most likely uniform near the bubble column outlet. The figures show that ANFIS  
442 prediction method can estimate gas hold-up profile almost identical with CFD method towards  
443 the column center for various sparger diameters and bubble column heights.

444 Another important parameter in prediction of bubble column is TKE. When bubbles travel to the  
445 column surface, the pressure energy is converted to turbulent kinetic energy. This parameter  
446 shows the intensity of turbulence inside the bubble column reactor. Fig. 14 (a-d) compares CFD  
447 simulation and ANFIS prediction of TKE versus radial position in the column. According to the  
448 figure, TKE towards the column center is higher than other region, while near the wall region it  
449 reaches to zero due to higher fluctuations of turbulent fluid velocities at the column center. As  
450 the sparger diameter increases, TKE reduces particularly towards the column center. According  
451 to the figures, for all ring sparger diameters, TKE near the sparger region is much higher than  
452 bubble column outlet.

453

454 Axial liquid velocity is the last parameter that has been estimated by ANFIS approach.  
455 The ANFIS prediction result is compared with CFD simulation results in Fig. 15(a-d). As the  
456 sparger diameter rises, the liquid centerline velocity reduces and results in flatter liquid velocity  
457 profile. Additionally, the centerline liquid velocity decreases, as the column height increases.  
458 This is attributed to the fact that, near the sparger region, swarm bubbles supply higher energy to  
459 the liquid than bulk region, and resulting in higher turbulent dissipation energy. This energy  
460 transferring shows the critical bubble column location where bubble plume breaks and bubbles  
461 split to smaller bubbles (break-up). In summary, the results show that ANFIS approach beside  
462 the CFD method is a capable prediction methodology to estimate the local hydrodynamics  
463 parameters at various column locations and operation conditions.

464 This section provides information about 3D bubble column hydrodynamics surface plots  
465 in order to understand the flow pattern and gas dynamics throughout the bubble column for all  
466 column heights and sparger diameters. The predicted gas hold-up contour has been presented for  
467 various column heights and sparger diameters in Fig. 16. This figure shows the predicted gas  
468 hold-up for sparger diameters of 0.085m (16(a)), 0.115m (16(b)) and 0.145m (16(c)). According  
469 to the figure, the gas hold-up profile has a non-uniform behavior towards the sparger, while at  
470 the middle and surface region (e.g., 1.3 and 2.5m) gas phase disperses uniformly. In general, the  
471 gas hold-up has a maximum value at the central region while the amount of gas decreases as  
472 sparger diameter increases around this region (see Fig. 16). Fig. 17 portrays the predicted axial  
473 liquid velocity in different column heights and sparger diameters. As seen, maximum liquid  
474 velocity occurs near the sparger region at almost every column height, while the liquid direction  
475 changes and results in two recirculation area near walls. Fig. 18 depicts the predicted TKE for  
476 various column heights and sparger diameters (0.085m (18(a)), 0.115m (18(b)) and 0.145m  
477 (18(c))). According to this figure, TKE towards the sparger is significantly higher than other  
478 regions. The figure shows that, with an increase in sparger diameter, TKE decreases.

479 The ANFIS method can predict the BCR hydrodynamics with different operational conditions in  
480 less computational time and provide continuous results. In order to examine the prediction  
481 ability, the BCR hydrodynamics are predicted for different column heights. All predicted results  
482 are compared with CFD results which are not used in training process.

483 The ANFIS method is used to predict the results of  $\epsilon_g$  at column heights (Y mesh coordinate) of  
484 0.43, 0.86, 1.3, 1.73, 2.16 and 2.56 m. The number of prediction data in X and Z mesh

485 coordinate is increased from 705 (CFD data) to 4800 nodes. For ANFIS training and model  
486 development, 70% of the actual data, which is the CFD results from benchmark case, at column  
487 heights of 0.217, 0.47, 0.73, 0.997, 1.3, 1.56, 1.8, 2.08, 2.3 and 2.6 m is given as an input. In  
488 addition, the number of data in X and Z mesh coordinate for the input is reduced to three quarter  
489 or to only 490 nodes. Please take note, the prediction is for the column heights that is not given  
490 as an input data to ANFIS model and hence the ability of the model is tested. Figure 19 shows  
491 the predicted contour plot of gas hold-up at various column heights (i.e., 0.432, 0.86, 1.3, 1.73,  
492 2.17 and 2.56 m) for ANFIS and CFD method. According to the figure, the ANFIS results are in  
493 good agreement with CFD results almost for all column heights. The ANFIS method predicts the  
494 circular gas hold-up distribution almost for all column heights, which is similar with CFD  
495 results. Both CFD and ANFIS show the higher gas hold-up at the center region of column,  
496 ranging 0.02-0.03, while this parameter reaches to zero value near the walls. Towards the sparger  
497 region ( $h=0.432\text{m}$ ), the sparger has influence on the gas distribution, resulting ring shape gas  
498 fraction (with 0.0012m inner and 0.09m outer diameter). As the column height increases, this  
499 influence diminishes and results in uniform distribution of gas. In comparison to CFD results,  
500 ANFIS method slightly over predicts the gas hold-up towards the walls at 2.56m. This may  
501 attribute to the fact that, ANFIS method cannot accurately recognize gas behaviour near the BCR  
502 boundary (particularly outlet). In order to enhance this over prediction, different ANFIS setting  
503 parameters or data filtering are required.

504 In summary, the combination of ANFIS and CFD prediction framework shows that, in case of a  
505 proper ANFIS learning process with CFD results, ANFIS approach can adequately predict  
506 bubble column hydrodynamics. In comparison to the CFD simulation, ANFIS approach provides  
507 the approximated bubble column hydrodynamics in a continuous domain. When the specific  
508 range of ring sparger diameters and column heights are trained in the ANFIS method, it can  
509 smartly approximate the flow pattern and gas dynamics within these particular ranges. On the  
510 other hand, in CFD simulation, the CFD simulation needs to be implemented for any changes in  
511 operation conditions due to production of discrete results. Therefore, providing a complete set of  
512 result for various conditions such as different sparger diameter requires computational efforts,  
513 resulting in large computational time.

514 In the last portion of this section, we propose the mathematical models of the liquid flow pattern  
515 and gas dynamics that have been estimated by selected ANFIS models. These mathematical

516 correlations show the axial liquid velocity, gas hold-up and TKE at different column heights,  
 517 radiuses and sparger diameters. The method of gaining mathematical models from ANFIS  
 518 approach was described in section 2.4. The formula of the relationship between gas hold-up and  
 519 effective variables, which are sparger diameter, axial coordination and radial coordination, can  
 520 be written as,

$$521 \quad \varepsilon_g = \frac{\sum_{i=1}^4 \sum_{j=1}^4 \sum_{k=1}^4 (\mu_{1i} \times \mu_{2j} \times \mu_{3k}) \times (p_m D_s + q_m H + r_m x + s_m)}{\sum_{i=1}^4 \sum_{j=1}^4 \sum_{k=1}^4 (\mu_{1i} \times \mu_{2j} \times \mu_{3k})} \quad (18)$$

522

523

in which

$$524 \quad \mu_{1i} = \frac{1}{1 + e^{-a_{1i}(x-c_{1i})}} - \frac{1}{1 + e^{-a_{2i}(x-c_{2i})}} \quad \mu_{2j} = \frac{1}{1 + e^{-a_{1j}(x-c_{1j})}} - \frac{1}{1 + e^{-a_{2j}(x-c_{2j})}}$$

$$525 \quad \mu_{3k} = \frac{1}{1 + e^{-a_{1k}(x-c_{1k})}} - \frac{1}{1 + e^{-a_{2k}(x-c_{2k})}}$$

526 First subscription of  $\mu$  shows the input number and the second subscription depicts the MF  
 527 number. The values of MFs' parameters (premise parameters) are shown in Table 4. Subscription  
 528 of  $m$  represents the rule number ranged from 1 to 64. Table 5 shows the values of  $p_m$ ,  $q_m$ ,  $r_m$ , and  
 529  $s_m$  (consequent parameters). Each MF has four parameters and linear portion of every rule has  
 530 four parameters. As seen in Table 4, the values of  $a_1$  and  $a_2$  for all MFs of individual input are  
 531 approximately equal.

532

533 ANFIS model with 4-4-3 configuration and Bell-shaped MF is select among 81 ANFIS  
 534 structures for prediction of turbulent kinetic energy. This model can be represented  
 535 mathematically as,

$$TKE = \frac{\sum_{i=1}^4 \sum_{j=1}^4 \sum_{k=1}^3 (\mu_{1i} \times \mu_{2j} \times \mu_{3k}) \times (p_m D_s + q_m H + r_m x + s_m)}{\sum_{i=1}^4 \sum_{j=1}^4 \sum_{k=1}^3 (\mu_{1i} \times \mu_{2j} \times \mu_{3k})} \quad (19)$$

537 in which

$$\mu_{1i} = \frac{1}{1 + \left| \frac{x - c_i}{a_i} \right|^{2b_i}} \quad \text{and} \quad \mu_{1j} = \frac{1}{1 + \left| \frac{x - c_j}{a_j} \right|^{2b_j}} \quad \text{and} \quad \mu_{1k} = \frac{1}{1 + \left| \frac{x - c_k}{a_k} \right|^{2b_k}}$$

539 Table 6 depicts premise parameters and illustrates that the values of  $b$  for all MFs are pretty close  
 540 to each other while  $a$  and  $c$  values are different. Forty-eight rules were constructed on the basis  
 541 of the selected ANFIS structure and the values of  $p_m$ ,  $q_m$ ,  $r_m$ , and  $s_m$  parameters in these rules are  
 542 portrayed in Table 7.

543  
 544 Axial liquid velocity is another element predicted by ANFIS approach. The selected ANFIS  
 545 model to estimate the axial liquid velocity has the highest simplicity among chosen ANFIS  
 546 models for prediction of gas hold-up and TKE. The equation of this model as follows:

$$V_y = \frac{\sum_{i=1}^2 \sum_{j=1}^4 \sum_{k=1}^4 (\mu_{1i} \times \mu_{2j} \times \mu_{3k}) \times (p_m D_s + q_m H + r_m x + s_m)}{\sum_{i=1}^2 \sum_{j=1}^4 \sum_{k=1}^4 (\mu_{1i} \times \mu_{2j} \times \mu_{3k})} \quad (20)$$

548 in which

$$\mu_{1i} = e^{-\frac{(x-c_i)^2}{2\sigma_i^2}} \quad \text{and} \quad \mu_{1j} = e^{-\frac{(x-c_j)^2}{2\sigma_j^2}} \quad \text{and} \quad \mu_{1k} = e^{-\frac{(x-c_k)^2}{2\sigma_k^2}}$$

550 The values of premise and consequent parameters are shown in Tables 6 and 7. Overall, there are  
551 20 premise parameters and 128 consequent parameters that have been refined by using CFD  
552 simulation axial liquid velocity results. Providing these types of mathematical models assists in  
553 improving the knowledge of flow pattern (flow field) and gas dynamics for various operation  
554 conditions. In addition, this model can predict much smarter when the number of trained data  
555 increases as input parameters (i.e., bubble column dimension in X, Y and Z direction, superficial  
556 gas velocity, gas and liquid properties).

557

558

559

560

#### 561 **4. Conclusions**

562 This paper presents the combination of CFD and ANFIS to predict the 3D bubble column  
563 hydrodynamics for various ring sparger diameters at different bubble column heights. The  
564 Eulerian approach is used to simulate the liquid flow pattern and gas dynamics inside the 3D  
565 cylindrical bubble column reactor. The CFD simulation results are validated with existing  
566 numerical, experimental and mathematical correlations. After validation process, the CFD  
567 simulation results are used to train ANFIS approximation approach. The proper architectures of  
568 the ANFIS prediction models, in terms of number and type of MFs, are investigated to find the  
569 most accurate prediction model. Based on this model, the mathematical correlations for bubble  
570 column hydrodynamics are developed, as the sparger diameter and bubble column height change.  
571 The conclusions of this study are as follows:

- 572 • Both CFD and ANFIS prediction method show that the axial liquid velocity, turbulent  
573 kinetic energy and gas hold-up rise towards the column centre, while these parameters  
574 reach to zero value near the column walls for various gas sparger diameters and bubble  
575 column heights. The larger ring sparger diameter produces flatter gas hold-up profile in  
576 the bubble column cross-section. In addition, for all sparger diameters, the centreline  
577 velocity, gas hold-up and turbulent kinetic energy are higher near the sparger region.



- 578 • ANFIS approach can predict the bubble column hydrodynamics in a very short time and  
579 provide a non-discrete result, while the CFD simulation needs to be employed for any  
580 changes in operation condition.
- 581 • Evaluation of different ANFIS structures illustrates that the type and number of  
582 membership function significantly affect the precision of the prediction model.
- 583 • The ANFIS method contains a good ability to predict hydrodynamics parameters of  
584 bubble column reactor which are not used in the training process. This will show that,  
585 this method can be used as assistance tools together with CFD methodology to predict  
586 parameters and minimize computational efforts, and numerical repetition.

587

588

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808 **List of Tables:**

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**Table 1** the equations of MFs used in the ANFIS model

Membership Function	Equation
Bell-shaped	$\frac{1}{1 + \left  \frac{x - c}{a} \right ^{2b}}$
Gaussian	$e^{-\frac{(x-c)^2}{2\sigma^2}}$
Sigmoidal	$\frac{1}{1 + e^{-a_1(x-c_1)}} - \frac{1}{1 + e^{-a_2(x-c_2)}}$

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Table 2 RMSE of utilized ANFIS structures

Number of MF			Output	RMSE for			Number of MF			Output	RMSE for		
$D_s$	$H$	$x$		Bell-Shaped MF	Gaussian MF	Sigmoidal MF	$D_s$	$H$	$x$		Bell-Shaped MF	Gaussian MF	Sigmoidal MF
2	2	2	$\epsilon_g$	0.0059	0.0060	0.0060	3	3	4	$\epsilon_g$	0.0041	0.0042	0.0040
			TKE	0.0025	0.0026	0.0030				TKE	0.0016	0.0017	0.0017
			$V_v$	0.0261	0.0263	0.0278				$V_v$	0.0131	0.0146	0.0162
2	2	3	$\epsilon_g$	0.0049	0.0049	0.0051	3	4	2	$\epsilon_g$	0.0056	0.0057	0.0057
			TKE	0.0021	0.0019	0.0022				TKE	0.0017	0.0021	0.0020
			$V_v$	0.0201	0.0207	0.0226				$V_v$	0.0189	0.0210	0.0225
2	2	4	$\epsilon_g$	0.0046	0.0048	0.0048	3	4	3	$\epsilon_g$	0.0041	0.0043	0.0043
			TKE	0.0022	0.0018	0.0022				TKE	0.0015	0.0016	0.0016
			$V_v$	0.0187	0.0182	0.0213				$V_v$	0.0131	0.0143	0.0174
2	3	2	$\epsilon_g$	0.0059	0.0060	0.0059	3	4	4	$\epsilon_g$	0.0039	0.0038	0.0039
			TKE	0.0022	0.0023	0.0024				TKE	0.0015	0.0015	0.0016
			$V_v$	0.0223	0.0237	0.0239				$V_v$	0.0111	0.0122	0.0137
2	3	3	$\epsilon_g$	0.0047	0.0048	0.0047	4	2	2	$\epsilon_g$	0.0057	0.0058	0.0058
			TKE	0.0019	0.0019	0.0018				TKE	0.0024	0.0026	0.0027
			$V_v$	0.0166	0.0178	0.0194				$V_v$	0.0255	0.0254	0.0265
2	3	4	$\epsilon_g$	0.0046	0.0046	0.0044	4	2	3	$\epsilon_g$	0.0045	0.0047	0.0047
			TKE	0.0019	0.0018	0.0018				TKE	0.0023	0.0023	0.0023
			$V_v$	0.0149	0.0152	0.0172				$V_v$	0.0192	0.0196	0.0216
2	4	2	$\epsilon_g$	0.0058	0.0059	0.0059	4	2	4	$\epsilon_g$	0.0044	0.0045	0.0044
			TKE	0.0020	0.0020	0.0026				TKE	0.0022	0.0018	0.0020
			$V_v$	0.0205	0.0222	0.0217				$V_v$	0.0168	0.0176	0.0198
2	4	3	$\epsilon_g$	0.0044	0.0045	0.0047	4	3	2	$\epsilon_g$	0.0057	0.0057	0.0057
			TKE	0.0017	0.0018	0.0018				TKE	0.0017	0.0020	0.0023
			$V_v$	0.0139	0.0163	0.0177				$V_v$	0.0218	0.0226	0.0225
2	4	4	$\epsilon_g$	0.0043	0.0044	0.0040	4	3	3	$\epsilon_g$	0.0042	0.0045	0.0043
			TKE	0.0017	0.0017	0.0018				TKE	0.0016	0.0017	0.0016
			$V_v$	0.0127	0.0109	0.0152				$V_v$	0.0153	0.0163	0.0180
3	2	2	$\epsilon_g$	0.0057	0.0058	0.0058	4	3	4	$\epsilon_g$	0.0037	0.0040	0.0039
			TKE	0.0024	0.0026	0.0029				TKE	0.0015	0.0016	0.0014
			$V_v$	0.0255	0.0255	0.0280				$V_v$	0.0124	0.0144	0.0148
3	2	3	$\epsilon_g$	0.0045	0.0047	0.0047	4	4	2	$\epsilon_g$	0.0055	0.0057	0.0056
			TKE	0.0024	0.0024	0.0020				TKE	0.0016	0.0018	0.0022
			$V_v$	0.0193	0.0196	0.0223				$V_v$	0.0204	0.0209	0.0203
3	2	4	$\epsilon_g$	0.0044	0.0044	0.0045	4	4	3	$\epsilon_g$	0.0040	0.0043	0.0042
			TKE	0.0024	0.0024	0.0019				TKE	0.0013	0.0015	0.0015
			$V_v$	0.0173	0.0179	0.0203				$V_v$	0.0130	0.0144	0.0163
3	3	2	$\epsilon_g$	0.0057	0.0058	0.0058	4	4	4	$\epsilon_g$	0.0037	0.0034	0.0032
			TKE	0.0018	0.0022	0.0021				TKE	0.0013	0.0013	0.0013
			$V_v$	0.0223	0.0224	0.0232				$V_v$	0.0094	0.0104	0.0127
3	3	3	$\epsilon_g$	0.0043	0.0044	0.0043				$\epsilon_g$			
			TKE	0.0017	0.0019	0.0017				TKE			
			$V_v$	0.0157	0.0164	0.0189				$V_v$			

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**Table 3** CC of utilized ANFIS structures

Number of MF			Output	CC for			Number of MF			Output	CC for		
$D_c$	$H$	$x$		Bell-Shaped MF	Gaussian MF	Sigmoidal MF	$D_c$	$H$	$x$		Bell-Shaped MF	Gaussian MF	Sigmoidal MF
2	2	2	$\epsilon_g$	0.8997	0.8973	0.8984	3	3	4	$\epsilon_g$	0.9524	0.9499	0.9552
			TKE	0.9598	0.9414	0.9124				TKE	0.9845	0.9776	0.9838
			$V_y$	0.9272	0.9265	0.9171				$V_y$	0.9821	0.9778	0.9728
2	2	3	$\epsilon_g$	0.9331	0.9311	0.9282	3	4	2	$\epsilon_g$	0.9110	0.9063	0.9088
			TKE	0.9771	0.9499	0.9750				TKE	0.9767	0.9653	0.9639
			$V_y$	0.9576	0.9548	0.9459				$V_y$	0.9627	0.9536	0.9465
2	2	4	$\epsilon_g$	0.9399	0.9357	0.9342	3	4	3	$\epsilon_g$	0.9531	0.9487	0.9486
			TKE	0.9775	0.9505	0.9752				TKE	0.9840	0.9831	0.9830
			$V_y$	0.9635	0.9655	0.9522				$V_y$	0.9823	0.9786	0.9682
2	3	2	$\epsilon_g$	0.9008	0.8982	0.8993	3	4	4	$\epsilon_g$	0.9582	0.9601	0.9572
			TKE	0.9704	0.9673	0.9533				TKE	0.9850	0.9826	0.9852
			$V_y$	0.9473	0.9406	0.9398				$V_y$	0.9874	0.9847	0.9804
2	3	3	$\epsilon_g$	0.9379	0.9347	0.9373	4	2	2	$\epsilon_g$	0.9080	0.9047	0.9043
			TKE	0.9798	0.9800	0.9770				TKE	0.9596	0.9237	0.9309
			$V_y$	0.9711	0.9669	0.9605				$V_y$	0.9310	0.9315	0.9252
2	3	4	$\epsilon_g$	0.9413	0.9404	0.9460	4	2	3	$\epsilon_g$	0.9422	0.9378	0.9386
			TKE	0.9813	0.9798	0.9796				TKE	0.9422	0.9401	0.9339
			$V_y$	0.9770	0.9759	0.9691				$V_y$	0.9616	0.9599	0.9507
2	4	2	$\epsilon_g$	0.9025	0.8997	0.9016	4	2	4	$\epsilon_g$	0.9459	0.9437	0.9455
			TKE	0.9749	0.9680	0.9586				TKE	0.9780	0.9490	0.9441
			$V_y$	0.9558	0.9480	0.9504				$V_y$	0.9707	0.9677	0.9589
2	4	3	$\epsilon_g$	0.9460	0.9432	0.9386	4	3	2	$\epsilon_g$	0.9081	0.9092	0.9064
			TKE	0.9811	0.9803	0.9783				TKE	0.9772	0.9740	0.9424
			$V_y$	0.9798	0.9724	0.9672				$V_y$	0.9498	0.9462	0.9465
2	4	4	$\epsilon_g$	0.9493	0.9465	0.9556	4	3	3	$\epsilon_g$	0.9511	0.9430	0.9491
			TKE	0.9822	0.9814	0.9801				TKE	0.9843	0.9812	0.9818
			$V_y$	0.9832	0.9877	0.9760				$V_y$	0.9757	0.9723	0.9660
3	2	2	$\epsilon_g$	0.9067	0.9026	0.9037	4	3	4	$\epsilon_g$	0.9611	0.9547	0.9573
			TKE	0.9537	0.9217	0.9281				TKE	0.9915	0.9832	0.9900
			$V_y$	0.9306	0.9308	0.9157				$V_y$	0.9840	0.9785	0.9774
3	2	3	$\epsilon_g$	0.9422	0.9374	0.9372	4	4	2	$\epsilon_g$	0.9126	0.9080	0.9096
			TKE	0.9651	0.9499	0.9762				TKE	0.9802	0.9708	0.9475
			$V_y$	0.9611	0.9596	0.9477				$V_y$	0.9564	0.9542	0.9567
3	2	4	$\epsilon_g$	0.9457	0.9447	0.9440	4	4	3	$\epsilon_g$	0.9553	0.9489	0.9508
			TKE	0.9440	0.9463	0.9824				TKE	0.9917	0.9893	0.9870
			$V_y$	0.9687	0.9664	0.9568				$V_y$	0.9824	0.9786	0.9722
3	3	2	$\epsilon_g$	0.9082	0.9035	0.9056	4	4	4	$\epsilon_g$	0.9612	0.9687	0.9717
			TKE	0.9695	0.9574	0.9601				TKE	0.9936	0.9930	0.9915
			$V_y$	0.9476	0.9470	0.9433				$V_y$	0.9909	0.9888	0.9834
3	3	3	$\epsilon_g$	0.9494	0.9470	0.9473				$\epsilon_g$			
			TKE	0.9824	0.9791	0.9816				TKE			
			$V_y$	0.9742	0.9720	0.9625				$V_y$			

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**Table 4** Premise parameters of gas hold-up's ANFIS prediction model

Input	MF	$a_1$	$c_1$	$a_2$	$c_2$
Sparger Diameter	1	4.444e+02	5.495e-02	4.444e+02	7.955e-02
	2	4.444e+02	9.655e-02	4.444e+02	1.161e-01
	3	4.444e+02	1.067e-01	4.444e+02	1.333e-01
	4	4.444e+02	1.571e-01	4.444e+02	1.750e-01
Axial Coordination	1	1.538e+01	-4.332e-01	1.539e+01	-6.853e-02
	2	1.539e+01	1.943e-01	1.538e+01	1.286e+00
	3	1.538e+01	1.197e+00	1.538e+01	2.172e+00
	4	1.538e+01	2.160e+00	1.538e+01	3.033e+00
Radial Coordination	1	1.390e+02	-1.919e-01	1.390e+02	-4.025e-02
	2	1.390e+02	-5.078e-02	1.390e+02	3.719e-02
	3	1.390e+02	3.270e-02	1.390e+02	6.373e-02
	4	1.390e+02	5.365e-02	1.390e+02	1.918e-01

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**Table 5** Consequent parameters of gas hold-up's ANFIS prediction model

Rule	$p$	$q$	$r$	$s$	Rule	$p$	$q$	$r$	$s$
1	2.06e-01	-9.83e-02	-3.13e-01	-4.81e-02	33	3.82e-01	2.60e-01	8.93e-01	4.96e-02
2	-5.34e-01	1.29e+00	-1.77e+00	4.37e-02	34	-6.72e-01	-1.27e-02	6.73e-01	8.78e-02
3	1.55e+00	8.96e-01	-1.35e+01	6.23e-01	35	-1.65e+00	8.76e-01	6.47e+00	-5.24e-02
4	-5.06e-01	-1.17e-01	-4.00e-01	9.51e-02	36	2.24e-01	2.53e-02	1.13e-01	-4.47e-02
5	-5.94e-01	3.96e-03	1.23e-01	5.75e-02	37	-2.40e-02	1.98e-03	3.64e-01	4.88e-02
6	-3.63e-01	7.86e-03	3.59e-01	7.16e-02	38	-4.16e-01	-1.05e-02	8.52e-02	9.34e-02
7	2.50e+00	2.43e-02	6.92e-01	-1.90e-01	39	8.82e-01	6.61e-03	1.75e-01	-9.68e-02
8	-6.96e-02	8.56e-03	-1.13e-01	1.52e-02	40	-1.48e-01	5.68e-03	-2.07e-01	4.29e-02
9	-3.33e-01	3.06e-03	2.04e-01	4.73e-02	41	-9.09e-02	1.57e-03	2.63e-01	4.72e-02
10	5.32e-01	-5.28e-03	1.87e-01	1.17e-02	42	-2.72e-01	-4.60e-03	3.95e-02	7.16e-02
11	1.43e+00	3.94e-03	3.02e-01	-9.52e-02	43	2.12e-01	-7.98e-04	7.83e-02	-5.00e-03
12	2.25e-01	4.51e-03	-2.32e-01	8.20e-03	44	2.08e-02	3.11e-03	-2.27e-01	2.46e-02
13	-2.26e-01	9.50e-03	2.00e-01	2.31e-02	45	-3.84e-02	5.75e-03	2.29e-01	2.65e-02
14	2.48e-01	3.12e-03	1.37e-01	1.05e-02	46	-2.20e-01	-5.82e-03	5.25e-02	6.97e-02
15	9.01e-01	-3.44e-03	2.06e-01	-3.86e-02	47	9.88e-02	3.85e-03	5.65e-02	-1.62e-03
16	1.43e-01	5.88e-03	-2.23e-01	8.75e-03	48	4.17e-02	6.62e-03	-2.09e-01	1.09e-02
17	5.02e-01	1.48e-02	4.36e-02	-4.27e-02	49	-6.29e-02	2.26e-01	9.12e-01	1.15e-01
18	-1.14e+00	6.72e-01	-1.65e+00	9.57e-02	50	-3.45e-01	8.55e-02	6.66e-01	3.97e-02
19	-2.40e+00	4.16e-01	-1.50e+00	4.44e-01	51	-4.37e-01	-5.38e-02	-5.82e-01	5.43e-02
20	2.88e-01	1.11e-02	3.96e-01	-7.65e-02	52	-3.72e-02	1.73e-01	-9.34e-01	1.23e-01
21	-1.99e-01	5.30e-03	2.05e-01	4.28e-02	53	1.17e-02	5.71e-03	3.95e-01	4.61e-02
22	-6.42e-01	-4.30e-03	1.13e-01	1.10e-01	54	-5.77e-01	-1.10e-02	-1.41e-01	1.14e-01
23	9.91e-01	6.76e-03	7.56e-01	-1.05e-01	55	-1.13e+00	-3.36e-03	-2.71e-02	2.05e-01
24	6.10e-03	7.05e-04	-1.77e-01	2.19e-02	56	3.13e-01	6.17e-04	-3.96e-01	3.84e-03
25	-2.00e-01	2.86e-03	2.33e-01	4.68e-02	57	6.16e-02	2.01e-03	2.95e-01	3.02e-02
26	-1.51e-01	-6.19e-03	6.05e-02	6.00e-02	58	-3.37e-01	-2.57e-03	-4.69e-02	7.39e-02
27	4.69e-01	-2.02e-04	1.72e-01	-2.64e-02	59	-2.38e-01	-3.14e-03	-1.39e-01	7.15e-02
28	8.65e-02	1.51e-03	-2.22e-01	2.04e-02	60	1.64e-01	1.06e-03	-2.82e-01	1.50e-02
29	-1.21e-01	6.74e-03	2.17e-01	2.77e-02	61	6.03e-02	5.18e-03	2.35e-01	1.60e-02
30	-1.56e-01	1.43e-03	5.58e-02	4.38e-02	62	-2.50e-01	-6.52e-03	-2.41e-02	7.24e-02
31	2.01e-01	5.99e-03	1.40e-01	-1.50e-02	63	-1.22e-01	-5.83e-03	-1.27e-01	5.98e-02
32	4.97e-02	4.99e-03	-2.05e-01	1.43e-02	64	1.08E-01	4.57E-03	-2.37E-01	1.05E-02

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**Table 6** Premise parameters of turbulent kinetic energy's ANFIS prediction model

Input	MF	<i>a</i>	<i>b</i>	<i>c</i>
Sparger	1	3.393e-03	2.000e+00	8.292e-02
Diameter	2	6.082e-03	2.001e+00	1.227e-01
	3	1.200e-02	2.000e+00	1.804e-01
	4	3.643e-03	2.000e+00	1.551e-01
Axial Coordination	1	4.189e-01	1.996e+00	-2.915e-02
	2	4.684e-01	2.001e+00	8.275e-01
	3	4.452e-01	2.000e+00	1.731e+00
	4	4.316e-01	2.000e+00	2.604e+00
Radial Coordination	1	1.028e-01	2.000e+00	-1.210e-01
	2	1.036e-01	1.999e+00	9.861e-03
	3	1.078e-01	1.999e+00	1.299e-01

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829 **Table 7** Consequent parameters of turbulent kinetic energy's ANFIS prediction model

Rule	$p$	$q$	$r$	$s$	Rule	$p$	$q$	$r$	$s$
1	2.80e-01	-1.33e-02	6.51e-02	-1.84e-02	25	-1.19e+00	-1.45e-01	-7.61e-01	5.80e-02
2	-4.17e-01	1.94e-01	-1.85e-01	2.40e-02	26	2.03e+00	7.35e-02	-6.50e-01	-2.77e-01
3	1.07e-01	-4.24e-02	-2.82e-01	4.06e-02	27	-8.98e-01	6.99e-02	-7.17e-01	2.47e-01
4	1.60e-01	2.09e-04	6.26e-01	8.82e-02	28	-1.18e+00	-2.83e-03	-1.10e-01	1.56e-01
5	-8.39e-02	2.38e-03	3.88e-01	4.28e-04	29	-2.52e+00	1.10e-02	-1.51e-01	3.12e-01
6	-1.53e-01	-5.04e-03	-2.20e-01	3.61e-02	30	1.79e+00	-7.02e-03	-1.02e+00	-8.78e-02
7	7.07e-02	-1.14e-03	5.45e-01	8.36e-02	31	-1.05e-01	2.31e-03	1.94e-01	5.19e-02
8	2.65e-01	2.77e-03	2.38e-01	-3.78e-02	32	-3.10e+00	1.05e-02	4.82e-01	4.13e-01
9	-2.50e-01	-4.47e-03	-2.82e-01	6.38e-02	33	1.11e+00	-1.03e-02	-2.47e-01	-1.15e-01
10	-1.36e-02	-6.68e-04	3.57e-01	6.19e-02	34	1.80e-01	1.41e-02	1.36e-01	-3.22e-02
11	-2.73e-02	-1.92e-03	1.88e-01	3.26e-03	35	-6.42e-01	1.17e-02	1.98e-01	6.21e-02
12	-5.67e-02	7.82e-03	-1.84e-01	7.49e-03	36	8.92e-02	-2.20e-02	-1.53e-01	5.87e-02
13	1.78e-01	6.15e-03	3.50e-01	3.15e-02	37	8.98e-01	5.27e-02	6.83e-01	-3.04e-02
14	-7.65e-02	1.20e-01	1.18e-01	-1.25e-02	38	-1.75e+00	5.31e-02	3.79e-01	2.31e-01
15	3.50e-02	-1.32e-02	-2.33e-01	2.75e-02	39	1.27e+00	-1.36e-02	-1.27e-01	-1.84e-01
16	1.65e-01	-1.60e-03	7.61e-01	1.04e-01	40	9.84e-01	-4.15e-03	9.44e-01	1.32e-02
17	-2.55e-01	3.30e-03	3.92e-01	-3.91e-04	41	1.59e+00	3.29e-03	5.41e-01	-2.92e-01
18	9.51e-03	-3.17e-03	-3.84e-01	4.86e-02	42	-1.44e+00	-9.89e-03	-3.76e-01	2.67e-01
19	7.87e-02	-1.26e-03	5.14e-01	7.51e-02	43	9.84e-02	-3.64e-03	4.83e-01	7.11e-02
20	-1.78e-01	-2.31e-03	1.73e-01	3.81e-03	44	2.00e+00	9.41e-04	1.84e-01	-3.23e-01
21	-6.91e-02	-6.62e-04	-3.70e-01	6.18e-02	45	-9.52e-01	-4.91e-03	-2.55e-01	1.87e-01
22	2.01e-02	-2.73e-03	3.56e-01	6.24e-02	46	-3.14e-01	-1.11e-02	3.23e-01	1.28e-01
23	-2.22e-01	-1.68e-03	1.36e-01	1.70e-02	47	1.22e-01	9.00e-03	1.33e-01	-5.63e-02
24	3.18e-02	-1.55e-03	-2.62e-01	3.70e-02	48	6.67e-02	-5.35e-03	-1.83e-01	2.88e-02

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833 **Table 8** Premise parameters of axial liquid velocity's ANFIS prediction model

Input	MF	$\sigma$	$c$
Sparger	1	3.107e-02	8.611e-02
Diameter	2	1.996e-02	1.526e-01
Axial	1	2.215e-01	-1.589e-01
Coordination	2	4.978e-01	8.053e-01
	3	4.685e-01	1.798e+00
	4	9.646e-02	2.793e+00
Radial	1	5.538e-03	-1.434e-01
Coordination	2	8.356e-02	-7.177e-02
	3	6.069e-02	4.728e-02
	4	9.647e-02	3.347e-01

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**Table 9** Consequent parameters of axial liquid velocity's ANFIS prediction model

rule	p	q	r	s	rule	p	q	r	s
1	7.86e-01	-2.80e-02	5.72e+00	7.48e-01	17	4.05e+00	-2.13e-01	-7.61e+00	-1.70e+00
2	-1.31e+00	-2.58e-02	-9.81e-01	-1.61e-02	18	-6.29e+00	2.47e-01	1.38e-01	9.77e-01
3	5.63e-01	3.97e-01	-1.06e+00	2.45e-02	19	6.33e+00	1.36e-03	8.17e-01	-1.04e+00
4	1.44e+00	-1.61e+00	-1.14e+01	1.80e+00	20	-3.41e+00	-2.88e-01	-2.54e+00	8.45e-01
5	5.01e-02	-1.27e-02	-5.08e+00	-6.45e-01	21	1.01e+00	-8.13e-03	-4.67e+00	-6.91e-01
6	-3.00e-02	1.28e-02	1.71e+00	1.16e-01	22	-1.41e+00	1.74e-02	4.10e+00	6.10e-01
7	1.17e-01	7.43e-03	-2.49e+00	2.31e-01	23	1.46e+00	-2.23e-03	2.36e+00	-3.01e-01
8	1.71e+00	-7.45e-02	2.01e+01	-2.85e+00	24	2.63e+00	1.01e-01	7.99e+01	-1.27e+01
9	-6.69e-02	-7.13e-03	-1.14e+01	-1.54e+00	25	5.03e-01	-1.76e-03	-1.08e+01	-1.53e+00
10	1.34e-01	6.32e-03	1.89e+00	1.33e-01	26	-6.83e-01	5.31e-03	2.87e+00	3.74e-01
11	-1.31e-01	-3.75e-03	-1.18e+00	1.71e-01	27	5.11e-01	3.22e-03	1.15e+00	-9.72e-02
12	1.13e+00	-2.02e-02	2.77e+01	-4.11e+00	28	2.05e+00	4.98e-02	5.42e+01	-8.72e+00
13	3.08e-01	1.54e+00	6.78e+00	-3.15e+00	29	-1.40e+00	5.06e-01	7.71e+00	-1.30e-01
14	-1.05e-01	-1.78e+00	-2.62e+00	4.44e+00	30	1.50e+00	-4.56e-01	-3.53e+00	6.58e-01
15	-1.95e-01	-2.42e+00	2.93e+00	6.00e+00	31	-2.47e+00	1.35e-01	2.20e+00	-1.69e-01
16	-4.68e-01	1.01e+01	-2.75e+01	-2.23e+01	32	-1.12E+00	-6.44E-01	-3.55E+01	6.94E+00

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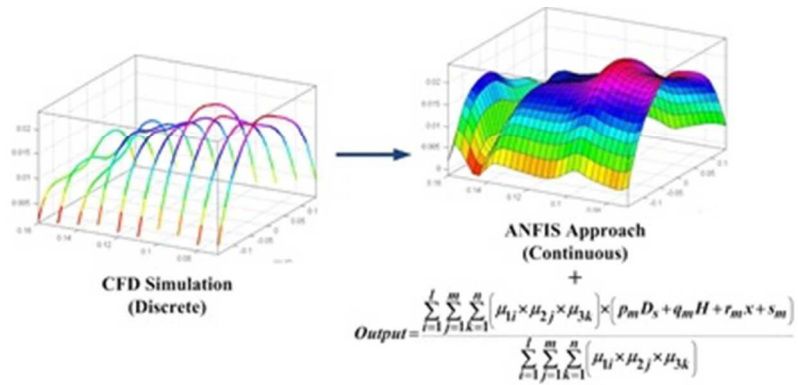
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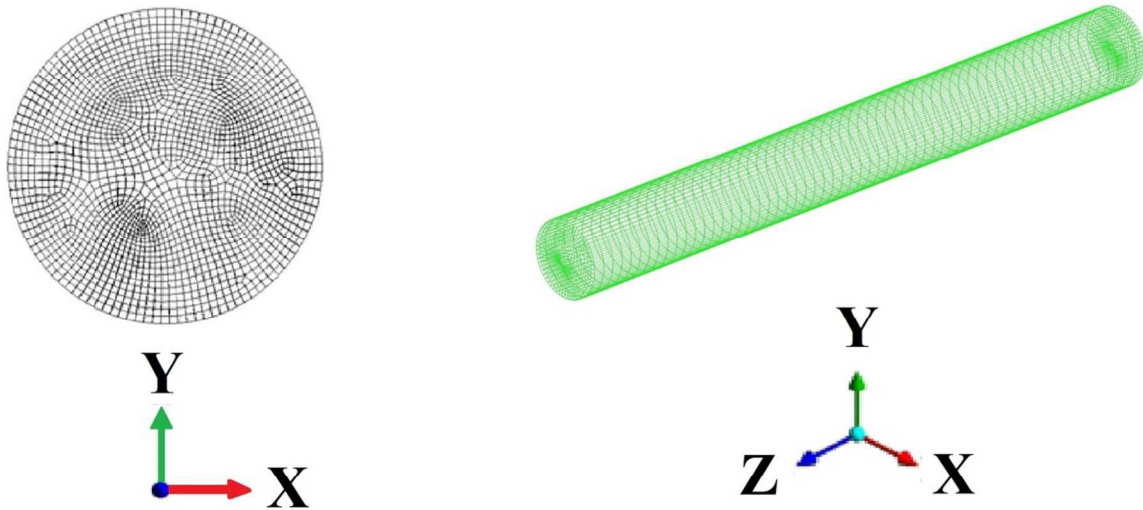
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853 **Figures**

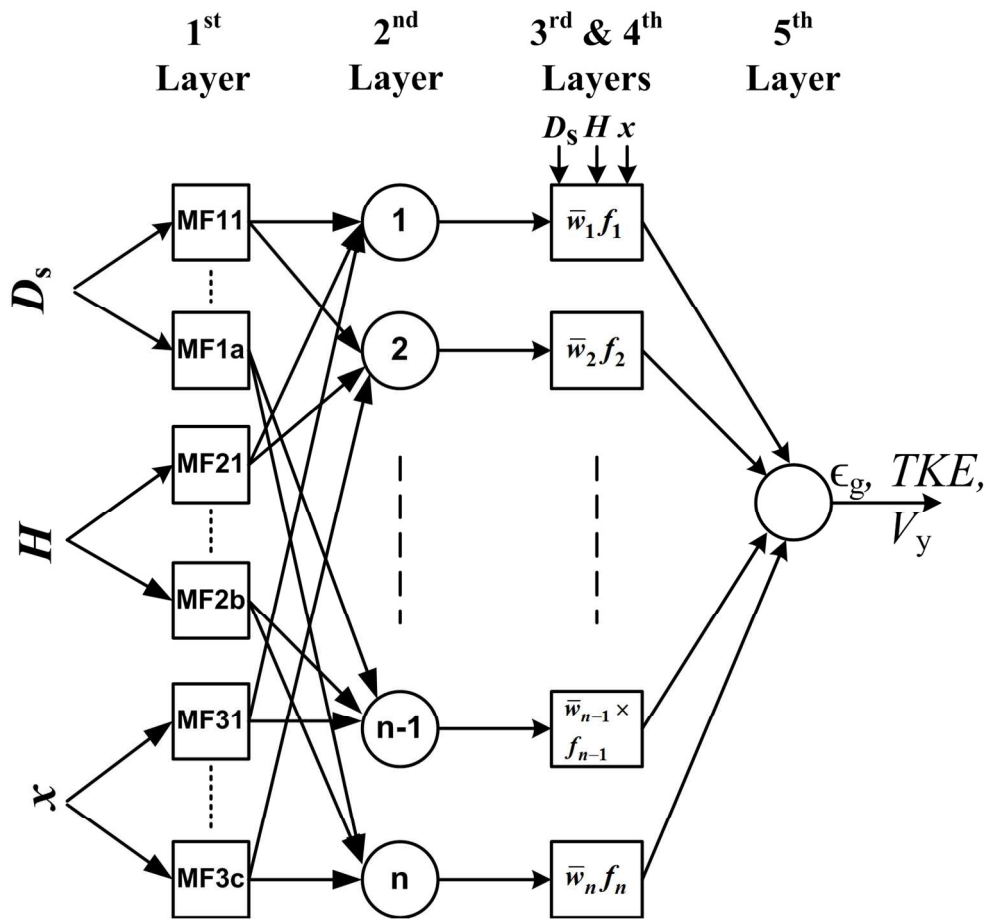
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855 **Figure 1** Schematic of prediction procedure of bubble column using the combination of CFD and ANFIS.

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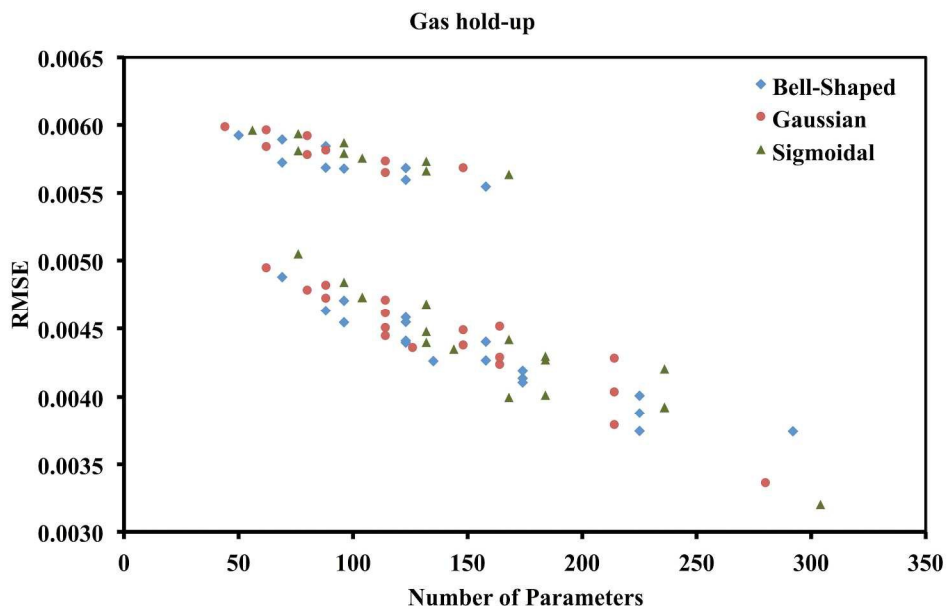
**Figure 2** Grid intensity of the present CFD study containing 40500 structural elements.



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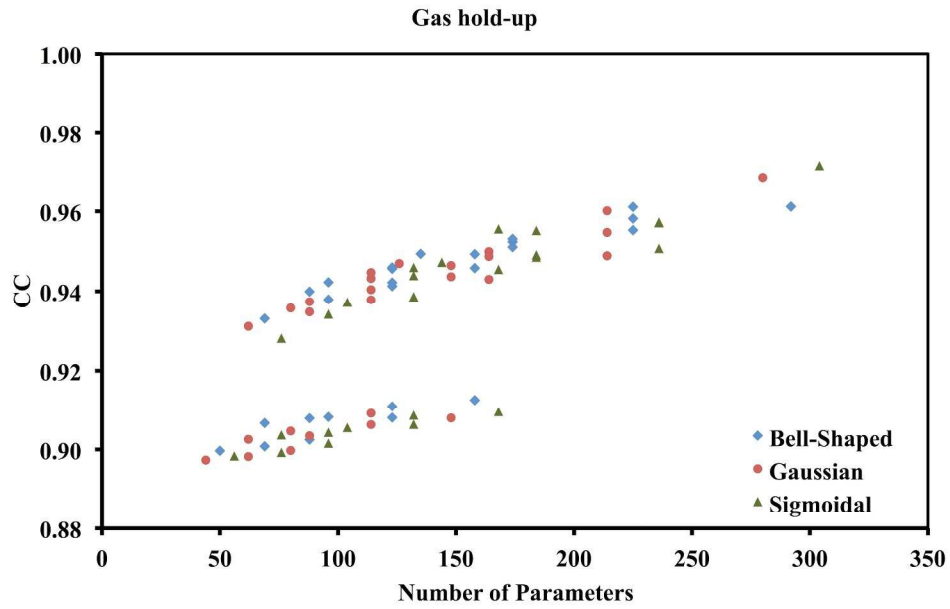
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Figure 3 Schematic of the ANFIS structure



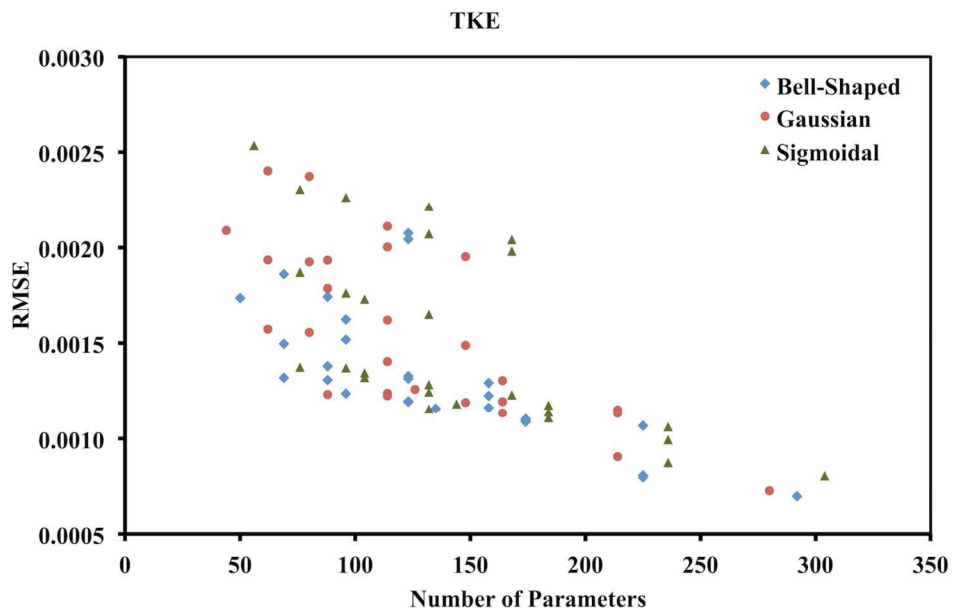
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Figure 4 Gas hold-up RMSE values of ANFIS methods versus number of parameters.



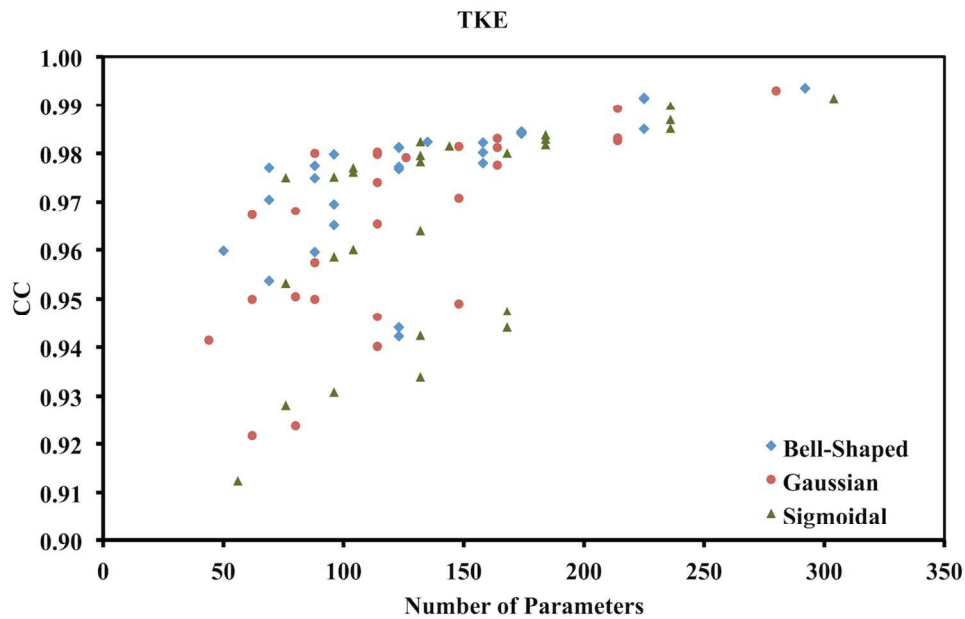
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**Figure 5** Gas hold-up CC of ANFIS methods versus number of parameters.



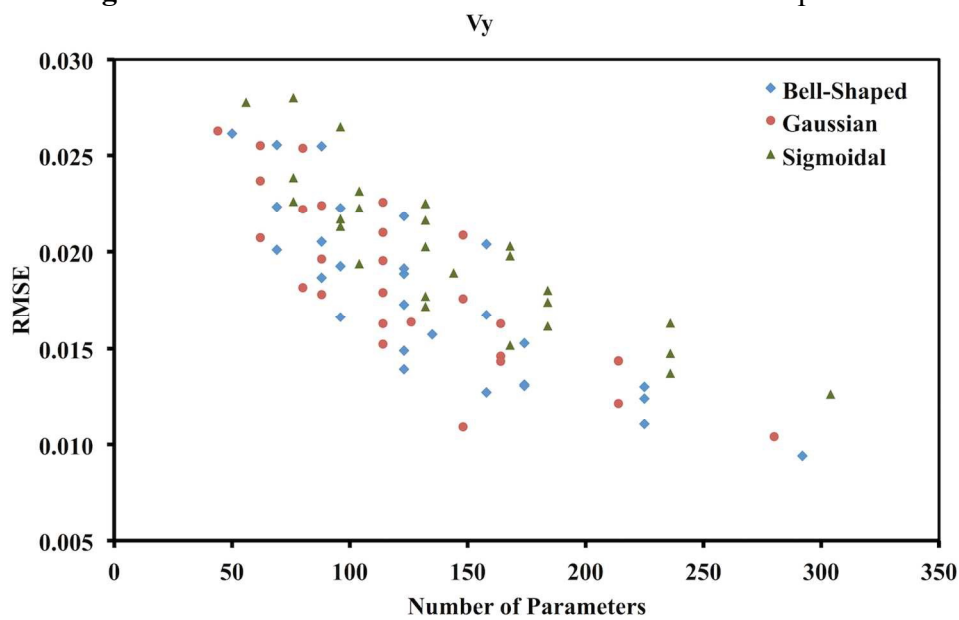
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**Figure 6** TKE RMSE values of ANFIS methods versus number of parameters.



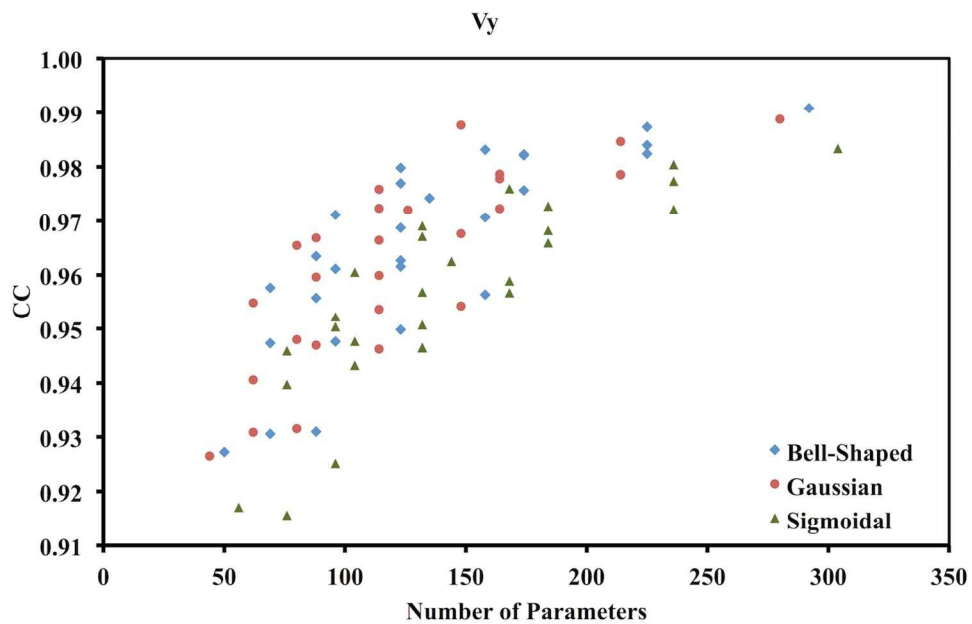
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**Figure 7** TKE CC of ANFIS methods versus number of parameters.



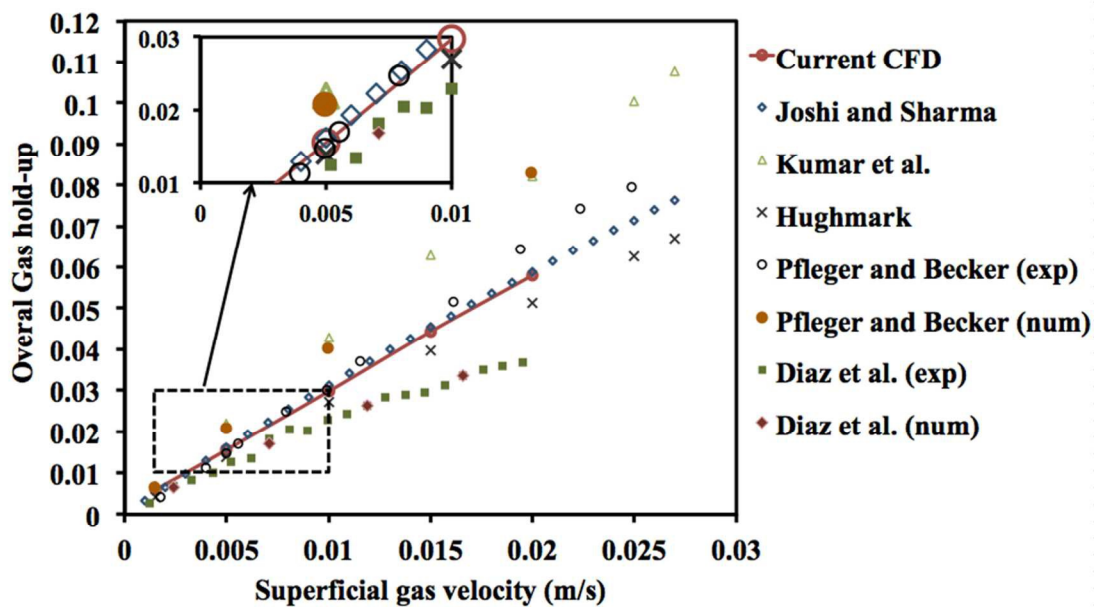
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**Figure 8** Axial velocity RMSE values of ANFIS methods versus number of parameters.



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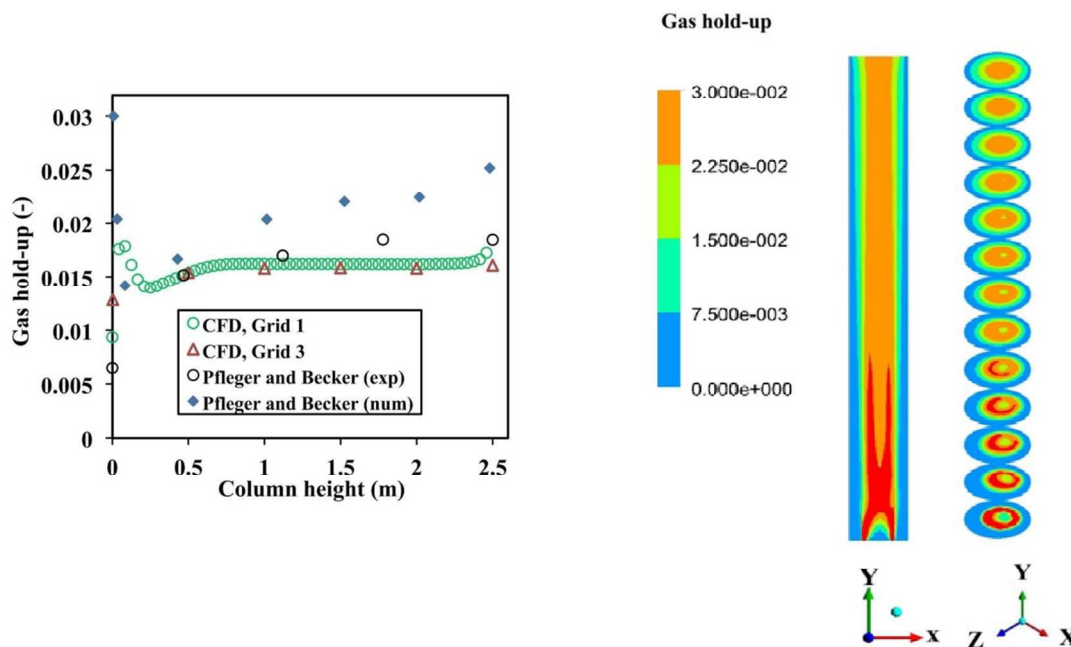
**Figure 9** Axial velocity CC of ANFIS methods versus number of parameters.



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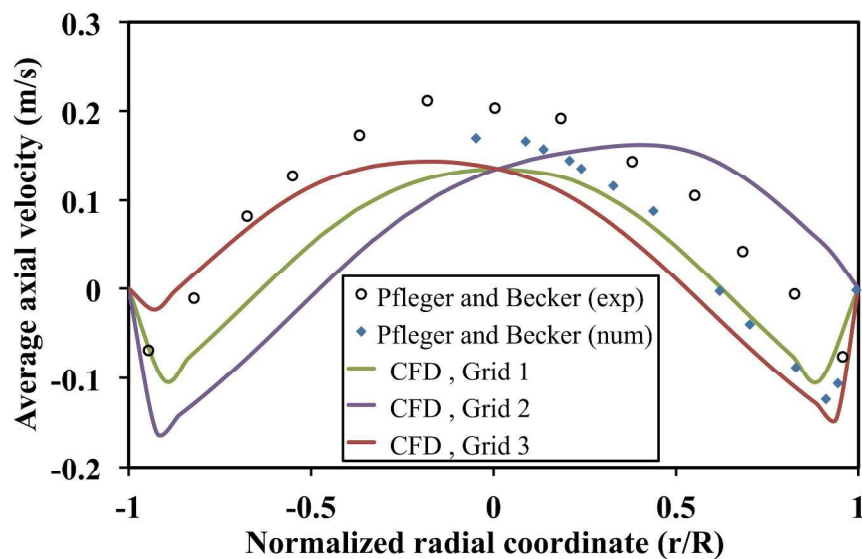
**Figure 10** Overall gas hold up for current simulation.





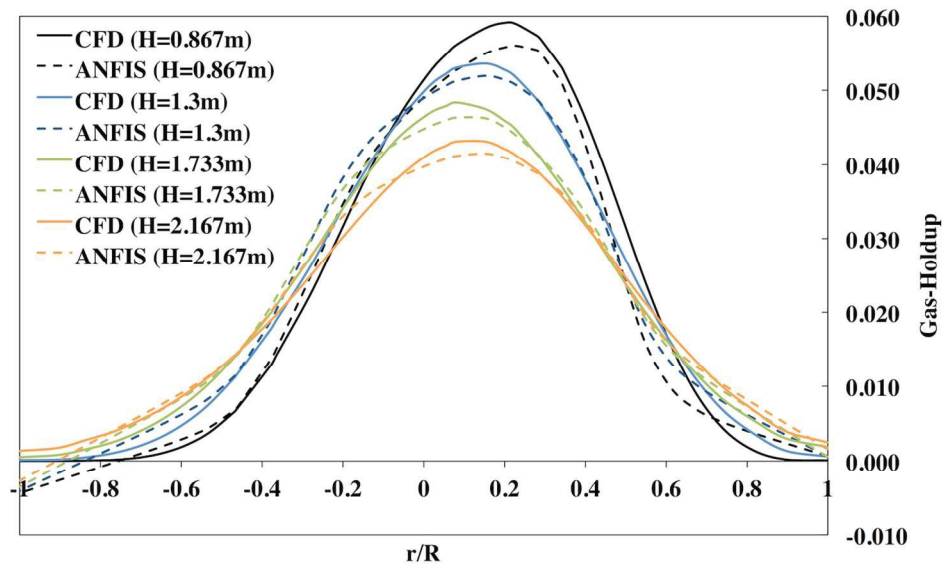
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875 **Figure 11** Overall averaged gas hold up for CFD results, Grids 1,3 and those from experiment  
 876 and simulation in Pflieger and Becker at various heights.



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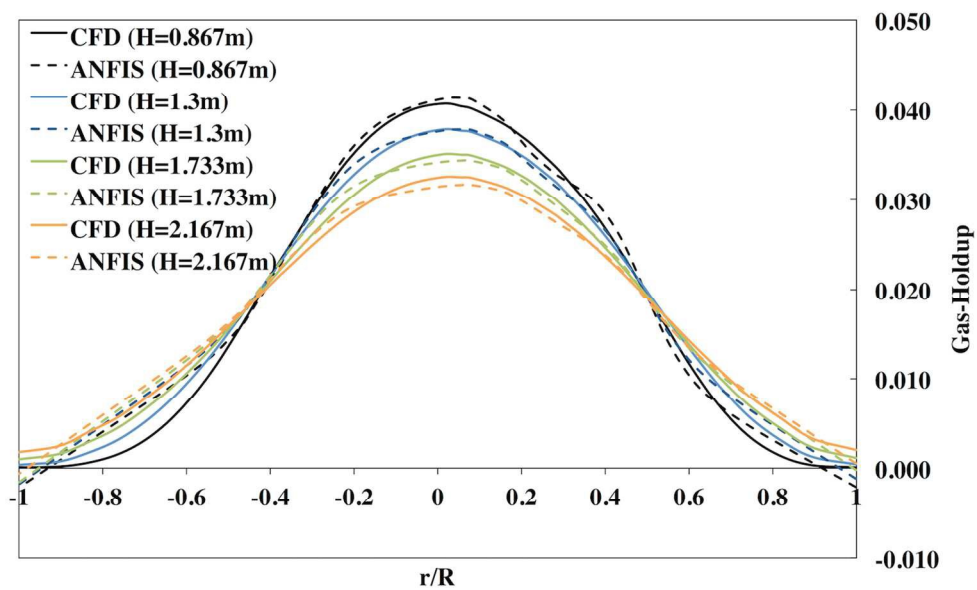
878 **Figure 12** Average axial liquid velocity for CFD, Grids 1-3 and those from experiment and  
 879 simulation in Pflieger and Becker at height 1.6m.



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**Figure 13(a)** Gas hold-up predicted and CFD values in sparger diameter of 0.08m.

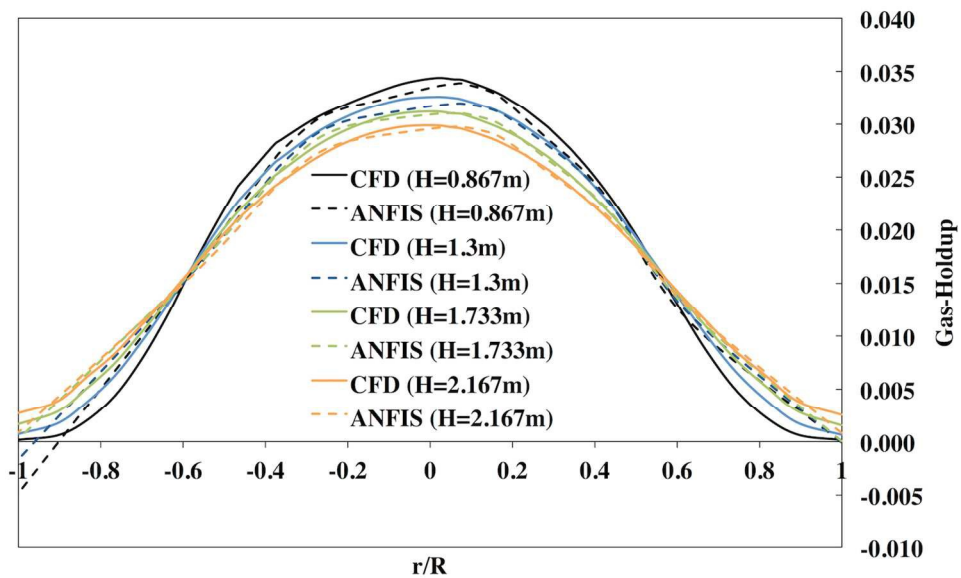


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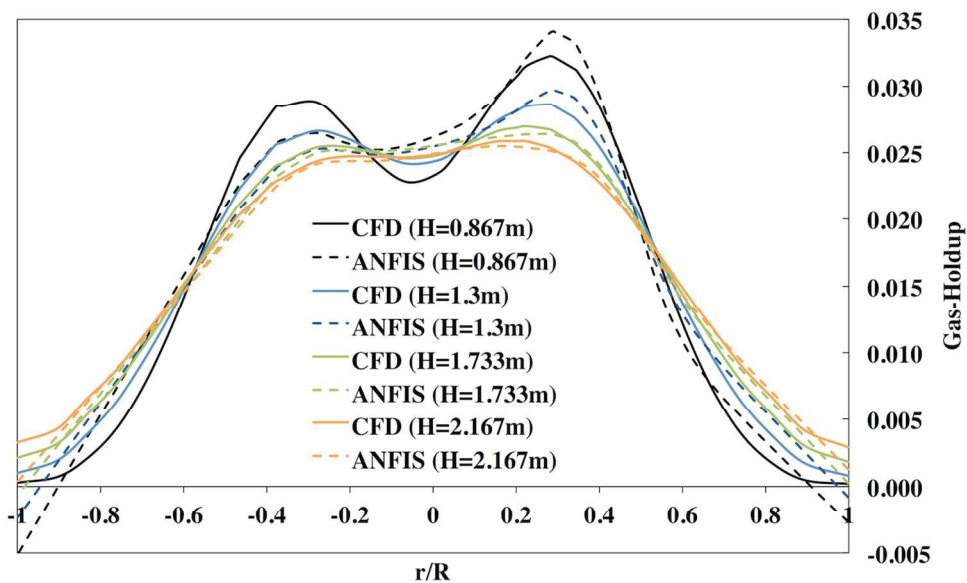
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**Figure 13(b)** Gas hold-up predicted and CFD values in sparger diameter of 0.10m.

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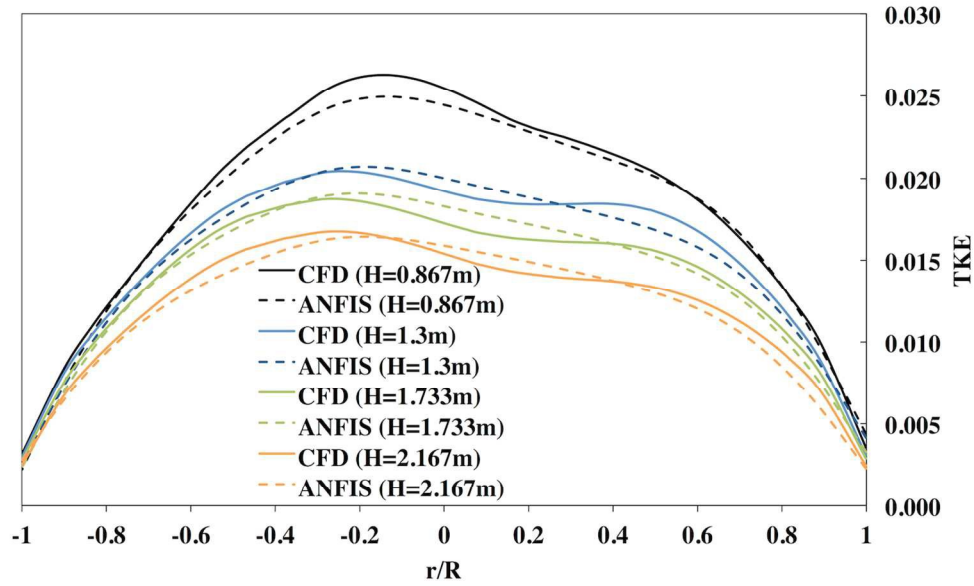


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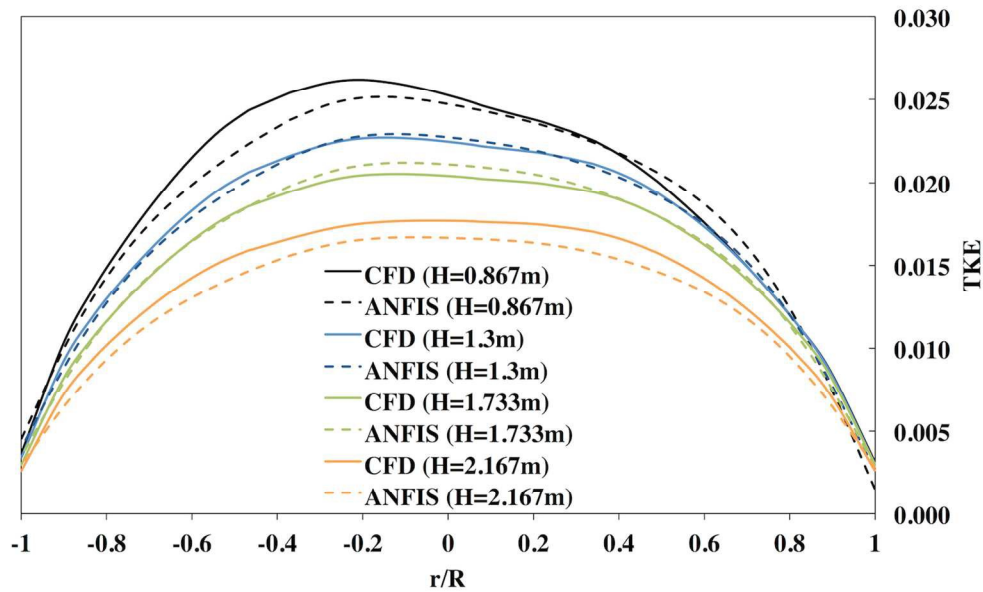
886 **Figure 13 (c)** Gas hold-up predicted and CFD values in sparger diameter of 0.12m.

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888 **Figure 13 (d)** Gas hold-up predicted and CFD values in sparger diameter of 0.14m.



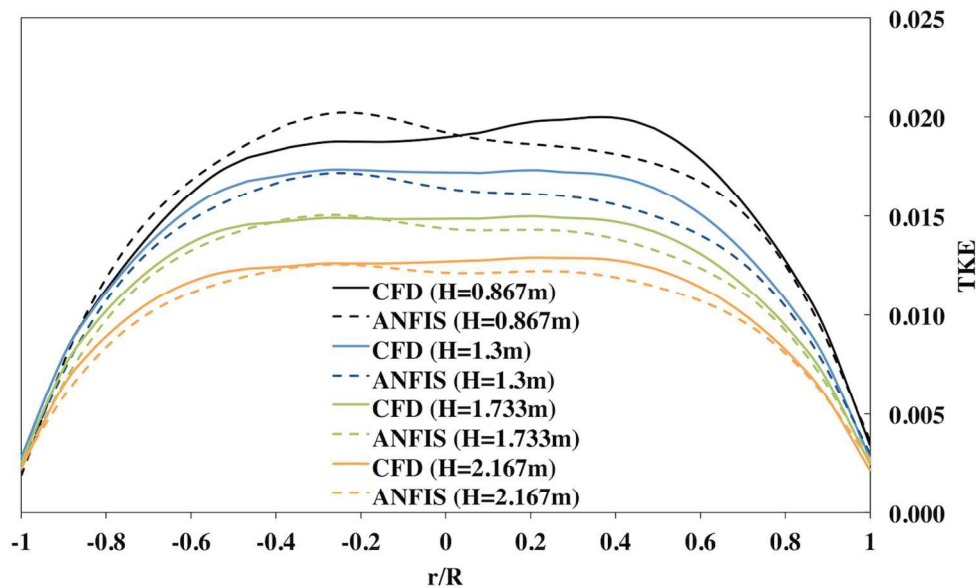
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890 **Figure 14 (a)** Turbulent kinetic energy predicted and CFD values in sparger diameter of 0.08m.

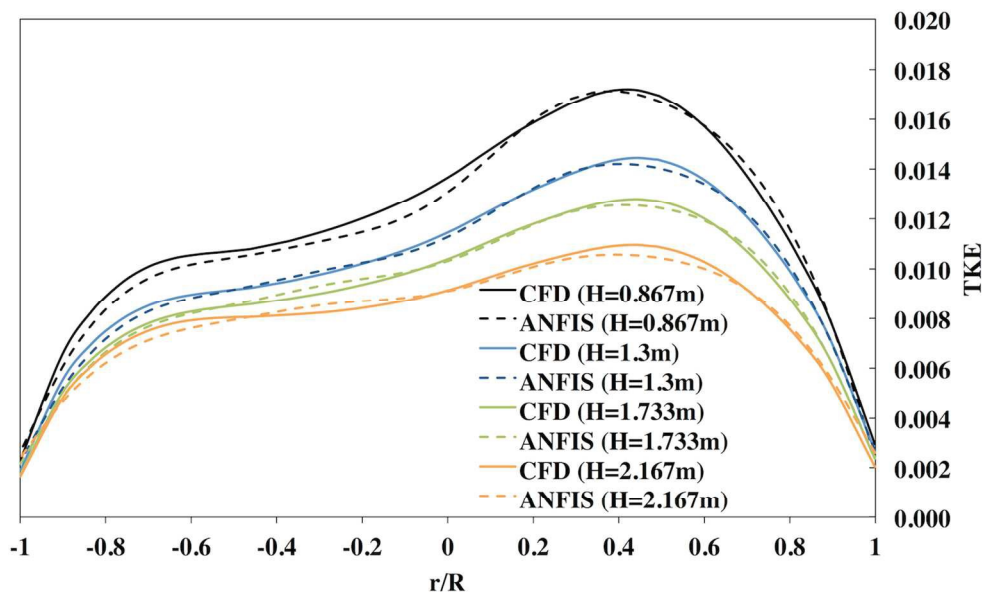
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892 **Figure 14 (b)** Turbulent kinetic energy predicted and CFD values in sparger diameter of 0.10m.

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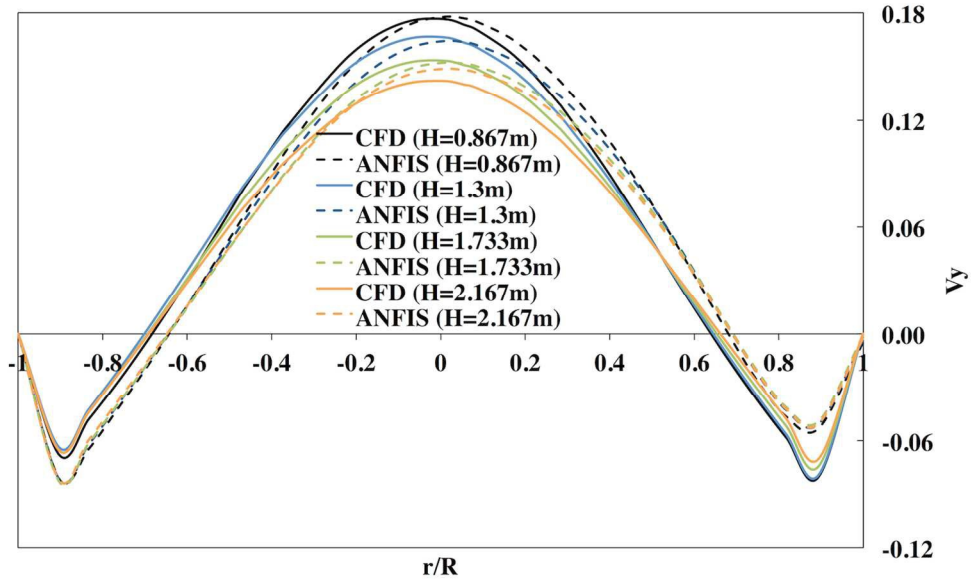


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895 **Figure 14 (c)** Turbulent kinetic energy predicted and CFD values in sparger diameter of 0.12m.

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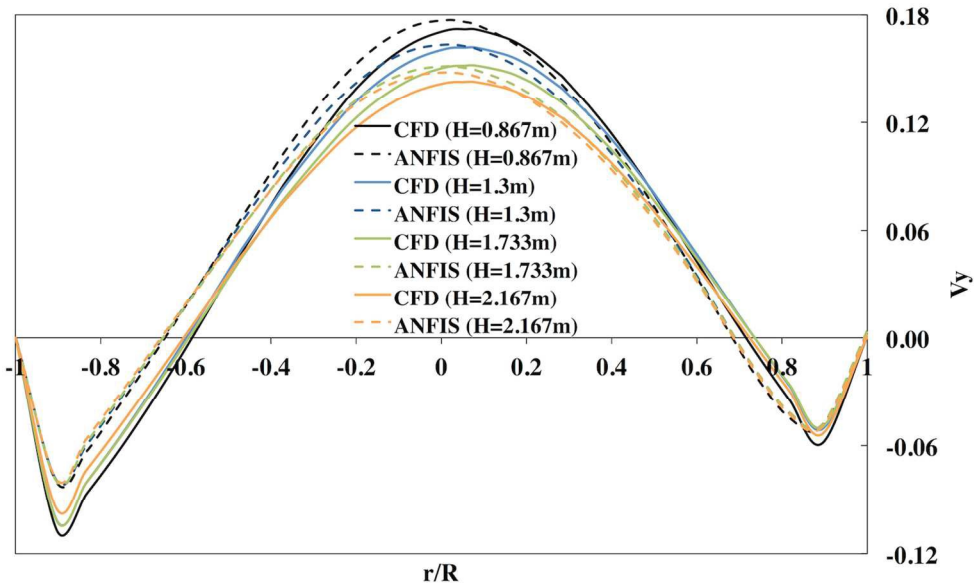
897 **Figure 14 (d)** Turbulent kinetic energy predicted and CFD values in sparger diameter of 0.14m.



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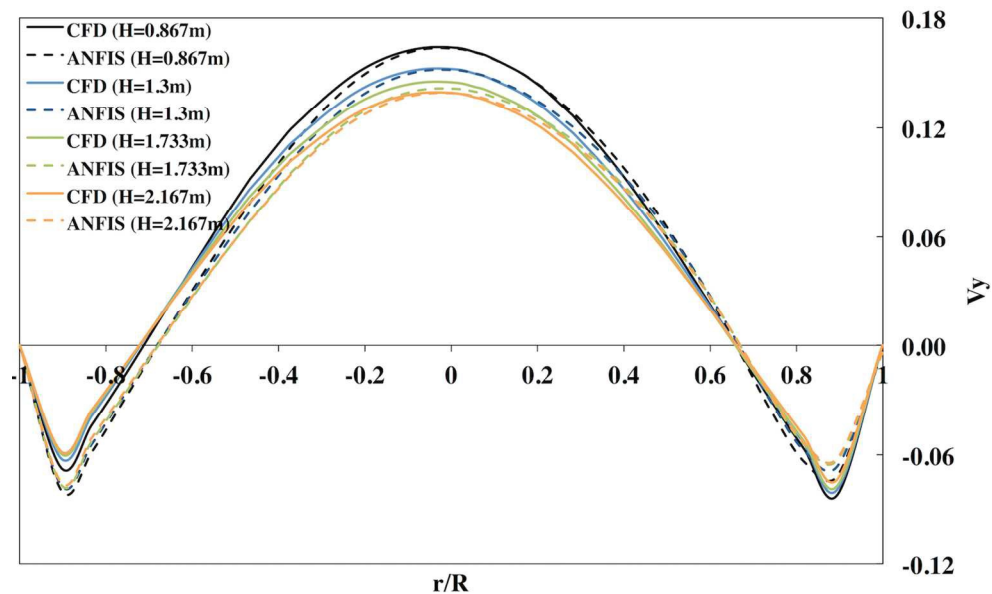
**Figure 15(a)** Axial velocity predicted and CFD values in sparger diameter of 0.08m.



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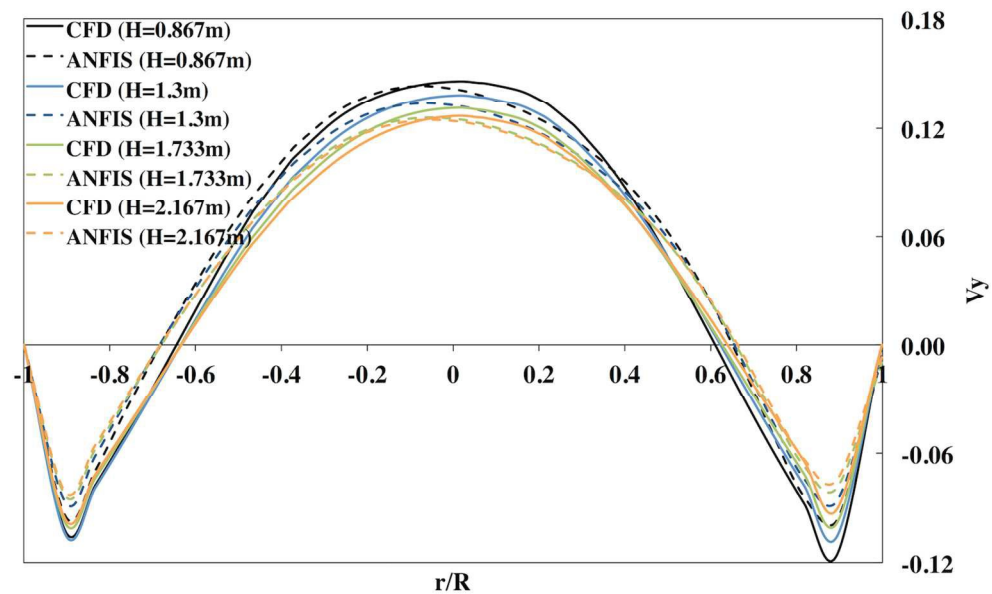
**Figure 15(b)** Axial velocity predicted and CFD values in sparger diameter of 0.10m.



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**Figure 15(c)** Axial velocity predicted and CFD values in sparger diameter of 0.12m.



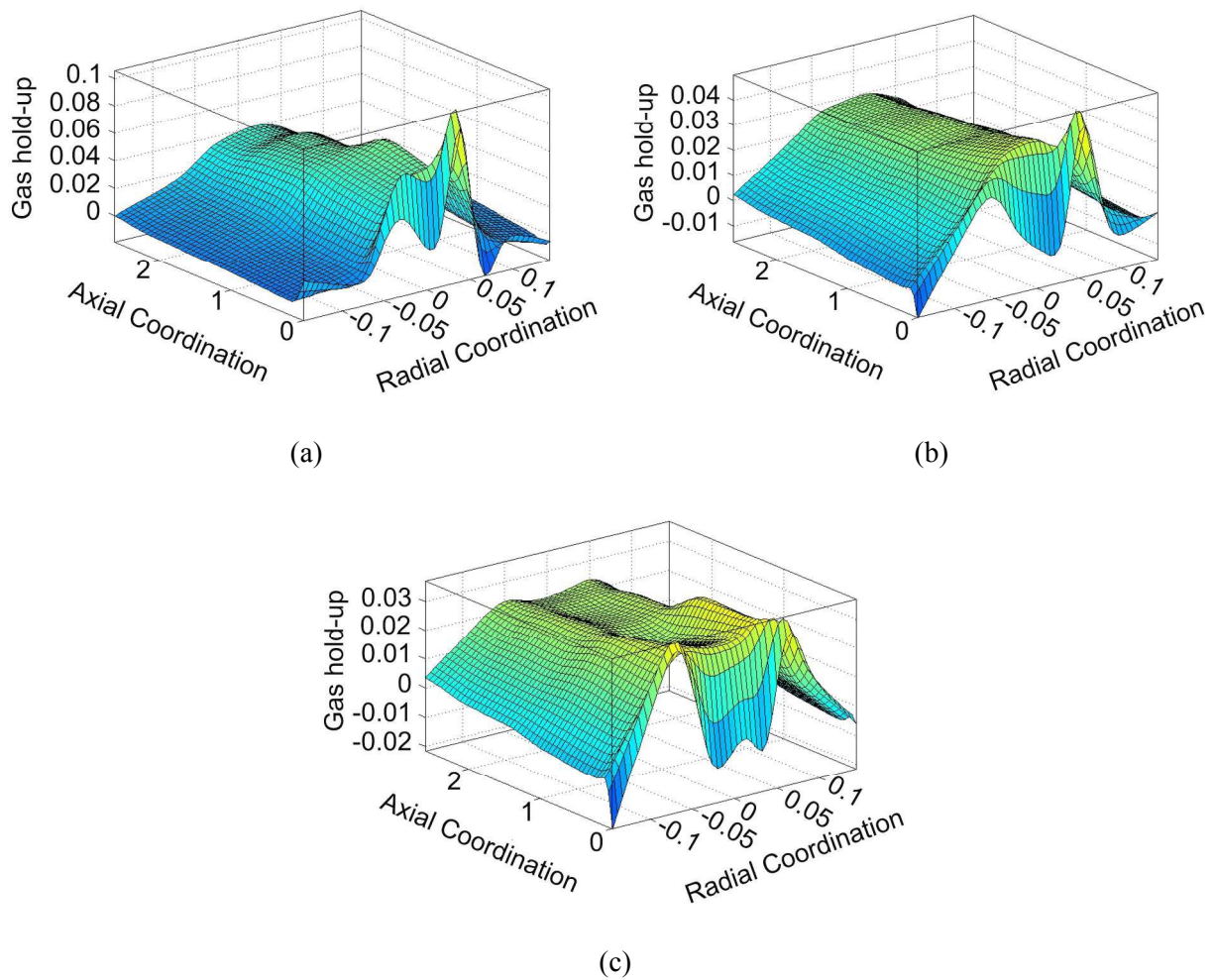
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**Figure 15(d)** Axial velocity predicted and CFD values in sparger diameter of 0.14m.



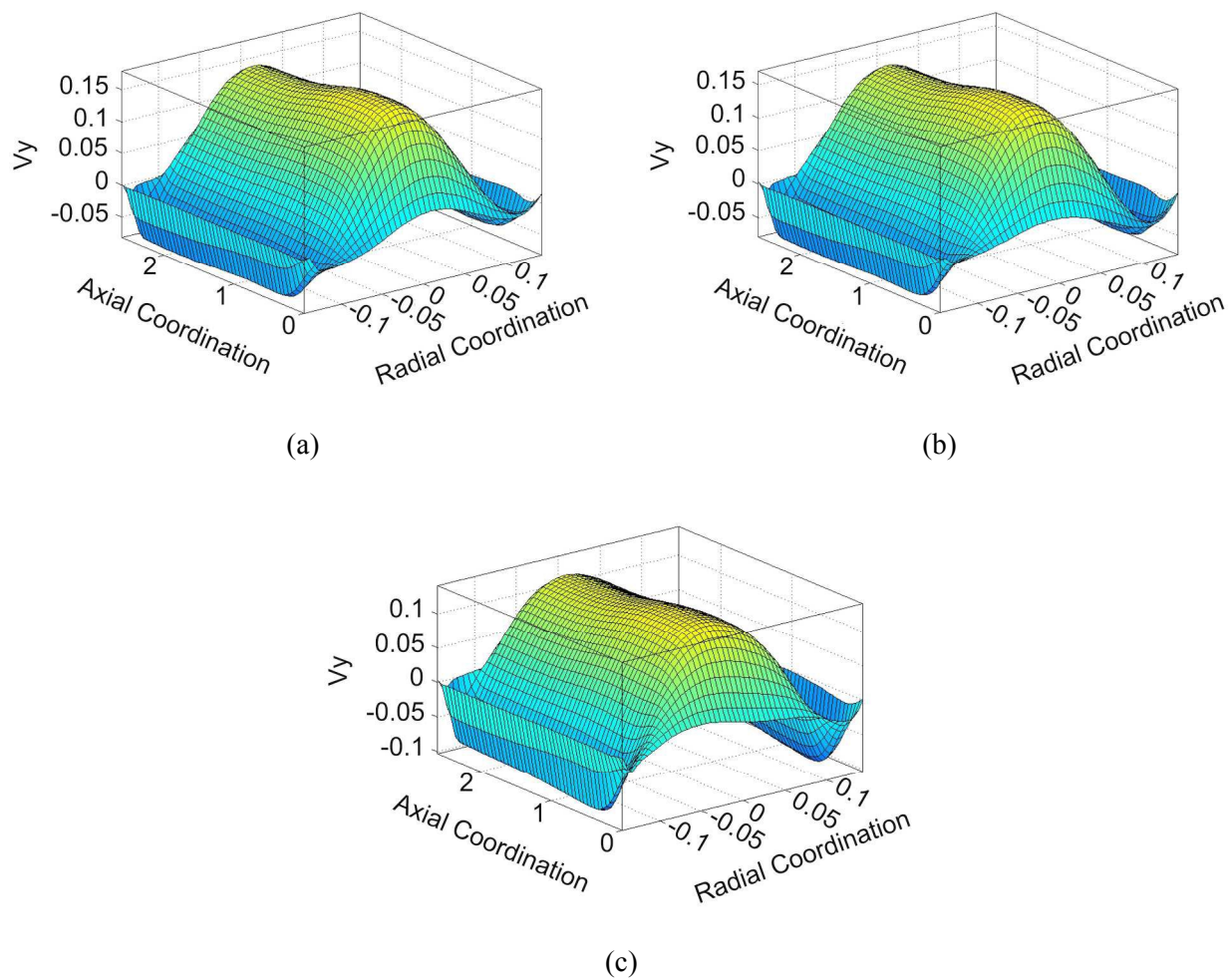


907 **Figure 16** Surface rules of selected ANFIS method in gas hold-up at sparger diameters of (a)  
908 0.085 (b) 0.115 and (c) 0.145.

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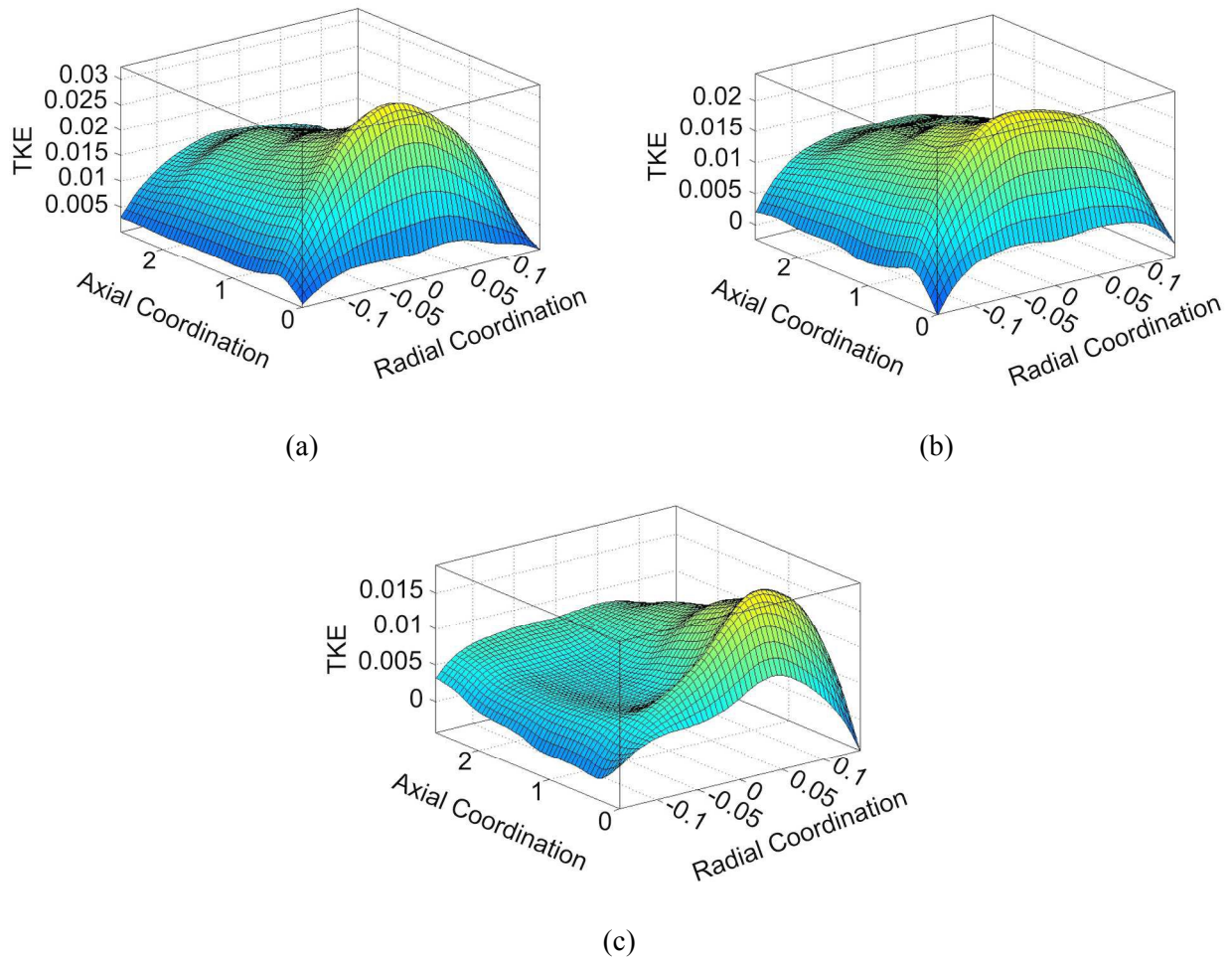
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911 **Figure 17** Surface rules of selected ANFIS method in axial velocity at sparger diameters of (a)  
912 0.085 (b) 0.115 and (c) 0.145.

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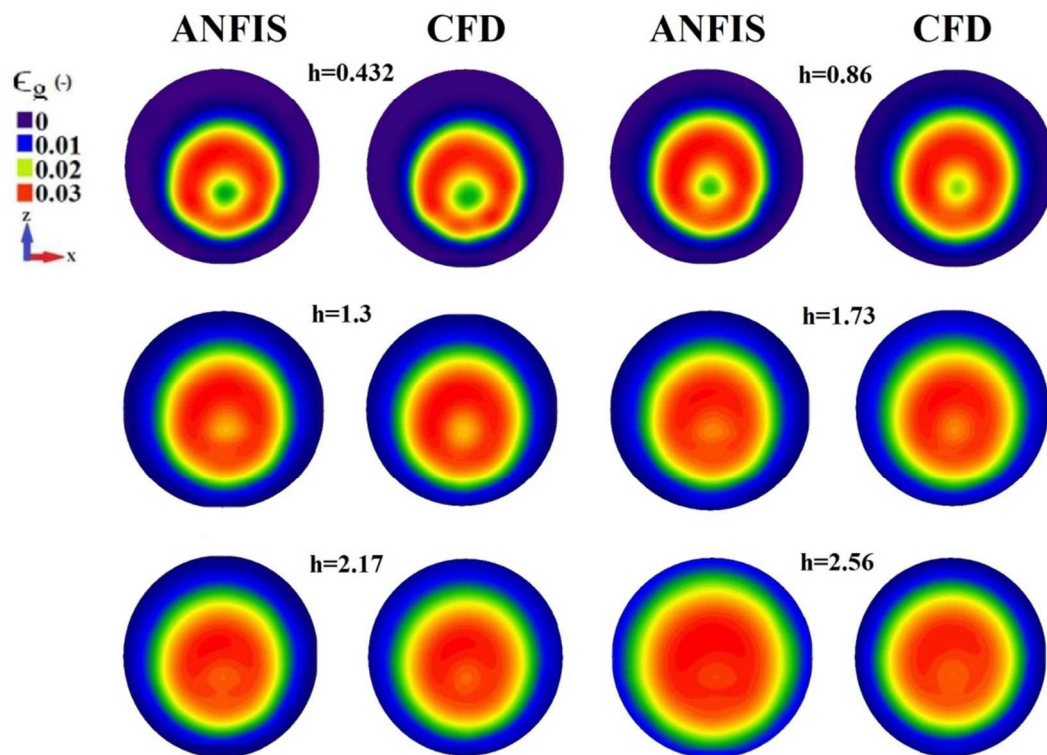
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915 **Figure 18** Surface rules of selected ANFIS method in turbulent kinetic energy at sparger  
916 diameters of (a) 0.085 (b) 0.115 and (c) 0.145.

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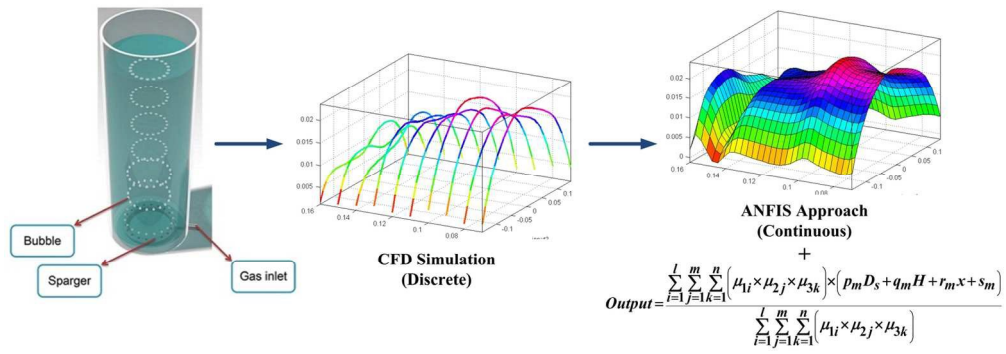
920 **Figure 19** Contour of gas hold-up at various column heights for ANFIS and CFD method.

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146x51mm (300 x 300 DPI)